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## NAVAL POSTGRADUATE SCHOOL Monterey, California





### **THESIS**

# THE DEVELOPMENT OF A MODEL BUILDER FOR A MICROCIRCUIT SUBSTRATE

by

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Thesis Advisor:

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## The Development of a Model Builder for a Microcircuit Substrate

by

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#### ABSTRACT

The Naval Postgraduate School is currently in possession of software designed to perform a thermal analysis of electronic components. This software package incorporates a model builder which contains two programs whose primary function is to generate a thermal model. In its present configuration, the model builder requires an inordinate amount of time for data input and model verification. This thesis describes the development of a model builder designed specifically to reduce the time required to model the substrate, epoxy and carrier layers of a microcircuit assembly.

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#### I. INTRODUCTION

The Naval Postgraduate School is currently in possession of software designed to perform a thermal analysis of electronic components. This software package incorporates a model builder which contains two programs whose primary function is to generate a thermal model, or input data file, to be read by the thermal analyzer program. The first program is considered to be a general model builder which is used in all model development stages as well as to modify an existing model. The second alternative was developed to generate a thermal model of a specific microcircuit geometry.

The development of an accurate thermal model of an electrical component requires that the structure be subdivided into a large number of small but finite subvolumes. Each subvolume is assumed to be isothermal with the centroids, also called nodes, considered to be representative of the entire subvolume. The most difficult problem encountered in the development of a thermal model is the generation of *n*-node equations in *n*-unknown temperatures where the nodes are connected by thermal conductances. As the desired accuracy of the thermal model increases, the number of required node equations becomes extremely large. Therefore, it is imperative

that the design engineer have access to a model builder that will produce the thermal model in a reasonable period of time.

In its present configuration, the thermal analysis software contains a model builder that generates the required node equations automatically. There is no question that the existing model builder programs have replaced the extremely laborious and time consuming process of generating the node equations by hand. However, they still require an inordinate amount of time for data input and model verification.

This thesis describes the development of a model builder designed specifically to reduce the time required to model the substrate, epoxy, and carrier layers of a microcircuit assembly. A typical microcircuit package configuration is shown in Figure 1. Figure 2a provides a horizontal interior illustration while Figure 2b displays the specific geometry to be modeled. All three layers may contain an equal number of nodes over their width. However, the carrier layer may contain a mounting ear on the front and rear surfaces. Additional characteristics to be discussed in what follows are:

- 1) The capability of working in English or SI units.
- 2) The choice of four aspect ratios.
- 3) The provision for up to 740 nodes depending on the existence of mounting surfaces (ears).
- 4) The ability to input heat dissipation using several methods.
- 5) The provision for six ambient temperatures.

- 6) The provision for rapid, menu-driven data input.
- 7) The automatic calculation of conductance values based on user input.

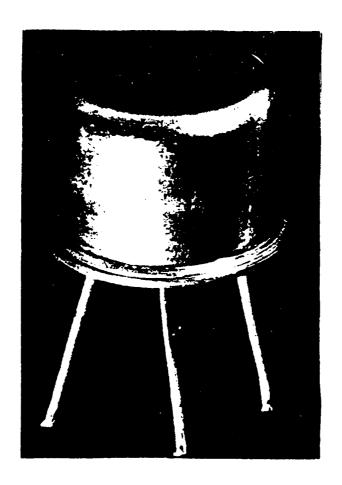
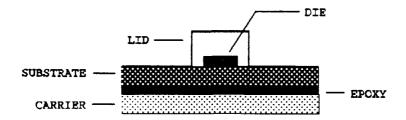
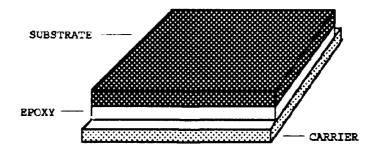


Figure 1. TO-5 configuration. (Courtesy of Honeywell, Inc.)



A. Horizontal cutout of a typical semiconductor assembly containing a single microcircuit die.



B. Area to be modeled.

Figure 2.

#### II. THE REASON FOR THERMAL ANALYSIS

Over the past several decades a trend of increasing sophistication and complexity has enveloped the electronics industry. This continuing advance in technology has greatly increased the reliability, capability, performance, and availability of electronic systems. The escalating demand for further advances in all areas of electronics has presented engineers with an abundance of complex problems.

One major area of concern is the continued development of advanced methods in the thermal control of multilayered structures. It is the responsibility of designers to ensure that electronic components operate efficiently and effectively throughout the specified thermal limits. Therefore, it is extremely important that design engineers have the capability to accurately and rapidly predict the temperature distribution on multilayered structures prior to prototype production. The overriding reasons for performing a precise thermal analysis are to increase component reliability, ensure proper material selection, ensure bias stabilization, and reduce or eliminate the possibility of catas\*rophic thermal failure.[Ref. 1]

#### A. RELIABILITY

There is a predictable relationship between the operating temperature of electronic components and reliability [Ref. 2]. The materials used in the fabrication of components have temperature limitations. Should these temperature boundaries be exceeded, the physical and chemical properties of the material are altered and the device fails. Figure 3 displays the intimate relationship between failure rate and component operating temperature for some selected devices. Furthermore, it is an established fact that the reliability of an electronic component is inversely proportional to the junction or component temperature and is also directly linked to failure rates [Ref. 1].

Consider Figure 4 which illustrates the "bathtub" mortality curve with the failure rate of a particular component plotted against component age during operation within thermal limits. The high failure rate in the interval prior to t<sub>b</sub>, also known as the burn-in period, is considered to be the result of poor quality control during the fabrication process.[Ref. 1]

The area of highest concern is the interval between  $t_b$  and  $t_{\star}$ . This period is considered to be the useful life, since with proper quality control, testing and burn-in procedures,  $t_b$  is equal to zero. Failures that occur in this interval are due to a variety of causes and are unpredictable. [Ref. 1]

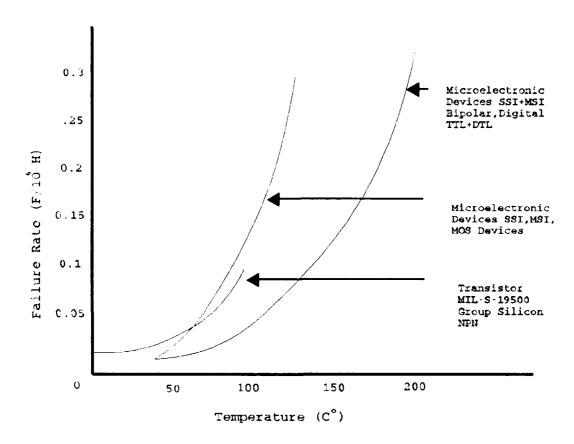


Figure 3. Failure rate vs. Temperature for selecte devices [Ref. 2].

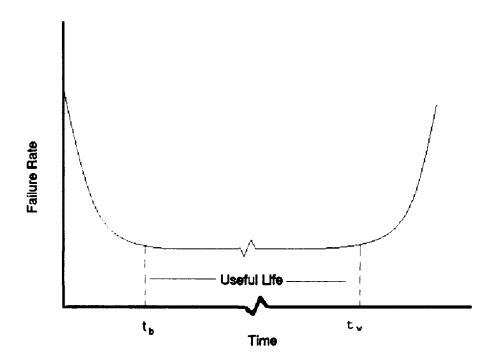


Figure 4. "Bathtub" mortality curve showing variation of failure rate with component age [Ref. 1].

Failure also occurs gradually due to sustained operations within specified temperature limits. If all aspects of fabrication have been performed correctly only a small fraction of components will have failed prior to reaching  $t_{\rm w}$ , also known as the wear-out period. Failures that occur in this period are due to the slow and never ending deterioration of materials.

#### B. MATERIAL SELECTION

The fabrication of electronic components results in the joining of several different materials. Consider Figure 2 which depicts a typical semiconductor structure. When power is applied to the components, heat is dissipated to the substrate and subsequently to the carrier. The mechanical properties of these materials are all affected differently by changes in temperature. Opposing mechanical and chemical reactions due to environmental conditions and contaminants may result in component performance degradation or a reduction in useful life. Table 1 lists temperature related factors that may affect component performance.

The primary objective in the selection of materials for the fabrication of an electronic assembly is to achieve the desired level of correlation within the finished product. As packaging densities increase thermal, mechanical, electrical, and chemical coupling becomes very strong. This high level of coupling can be both an advantage and a disadvantage. For example, a high level of correlation is desirable during fabrication to ensure an uniform product. However, in use, strong coupling is generally more desirable for moderate temperature deviations and weaker coupling is more desirable for large temperature deviations. In the case of large temperature deviations strong coupling may result in the catastrophic failure of many connected components while weak coupling may limit the number of failed components. Therefore, the strength of coupling between materials must be based on the type of failure most likely to occur and an accurate thermal analysis must supply this information to assist in proper material selection [Ref. 3].

TABLE 1. TEMPERATURE FACTORS [Ref. 4]

Mechanism	Effect on Equipment	Accelerating factors
Increasing Temperature	Loss of strength, reduced stiffness, reduced resonant frequency, softening, distortion, aging, and creep	Lubricants, rubber parts, plastics, corrosion, fatigue, load intensity, and time duration
Reducing Temperature	Increased viscosity, increased stiffness, increased resonant frequency, brittleness, and reduced impact resistance	Lubricants, rubber parts, plastics, and time duration
Thermal Expansion and Contraction	Change in size and shape, buckling, cracking, distortion, and loosening	Temperature cycling, temperature range, unequal expansion coefficients, stress concentrations , and lack of strain relief

#### C. BIAS STABILIZATION

The first step in the design and implementation of a semiconductor device is to establish a stable and predictable electrical operating point. This procedure, known as bias stabilization, attempts to determine a stable operating point that is virtually independent of external component parameters. However, as external parameters change, the operating point is directly affected. Therefore, a good bias design ensures that components will always operate within a certain range of their nominal value. [Ref. 5]

Consider Figure 5 which displays a transistor connected in the common-emitter configuration. Suppose that a proposed operation requires a specific collector to emitter voltage  $(V_{CE})$ . The circuit consists of a battery or some other source that provides a bias voltage  $V_{CC}$ , the collector resistor  $R_{C}$ , and the transistor. By Kirchoff's voltage law

$$-V_{CC} + R_C I_C + V_{CE} = 0 {1}$$

which results in a collector to emitter voltage of

$$V_{CE} = V_{CC} - I_{i}R_{C}$$
 (2)

Should the collector current be allowed to increase in excess of tolerable limits,  $V_{CE}$  must decrease because  $V_{CC}$  and  $R_{C}$  are

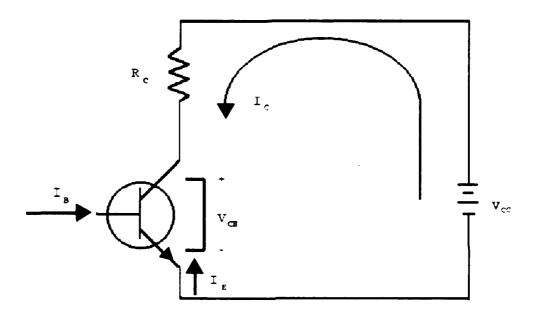


Figure 5. Schematic of transistor connected in common-emitter configuration [Ref. 1].

fixed values. Therefore, it should be noted that, if a high junction temperature causes an increase in  $I_{\text{C}}$ ,  $V_{\text{CE}}$  can no longer be maintained at the level desired to perform the desired operation. [Ref. 1]

#### 1. Operating in the Forward Bias Region

As an example of electronic component temperature dependence, consider a diode operating in the forward bias region. In the forward region the i-v relationship is closely approximated by

$$i = I_S(e^{\frac{v}{nV_T}} - 1) \tag{3}$$

In this equation  $I_s$  is a constant for a given diode at a given temperature. The current  $I_s$  is usually called the saturation current. However, another name for it is the scale current, which arises from the fact that  $I_s$  is directly proportional to the cross-sectional area of the diode. Furthermore, it can be seen in Table 2 that,  $I_s$  is a very strong function of temperature. [Ref. 5]

The temperature relationship between  $I_{\text{S}}$  and the forward current i is derived from the voltage  $V_{\text{T}}$ . This constant, called the thermal voltage, is given by

$$V_T = \frac{kT}{Q} \tag{4}$$

where

 $k = Boltzman's constant, 1.38 \cdot 10^{-23} J/ K$ 

T = the absolute temperature, K

 $q = the charge on the electron, 1.602 \cdot 10^{-19} C$ 

Table 2 illustrates this relationship and emphasizes the need to accurately analyze a proposed assembly prior to fabrication.

TABLE 2. TEMPERATURE DEPENDENCY OF  $\mathbf{I}_{S}$  ON  $\mathbf{i}$  FOR SELECTED MATERIALS

GERMANIUM			SILICON		
∙c	Is	i	Is	i	
25	3.0 μΑ	18.01 μA	50.0 ηA	82.31 ηA	
95	0.384 mA	1.473 mA	51.2 μA	61.39 μA	
165	49.2 mA	0.136 A	52.4 mA	49.2 mA	

#### D. CATASTROPHIC THERMAL FAILURE

Another of the primary goals of techniques in advanced thermal control is to provide a thermal environment for a diversity of components that are in increasingly close proximity to each other. Figure 6 illustrates the increasing level of packaging densities. With increasing complexity comes an increased level of connections between dissimilar material and a greater possibility for exceeding temperature limitations. Therefore, it is necessary that designers have

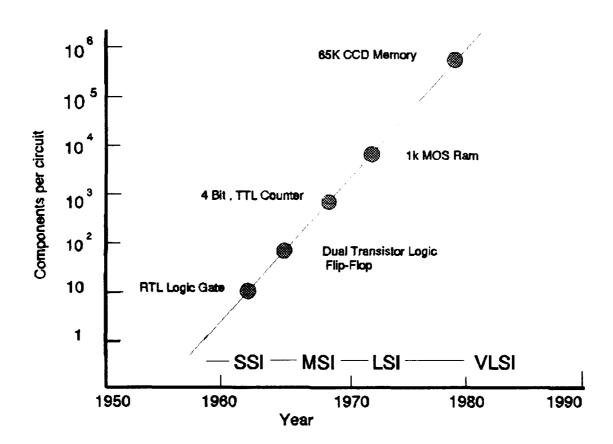


Figure 6. State of the art in circuit complexity [Ref. 1].

some knowledge of the possible range of component and environmental variations in order to prevent catastrophic thermal failure.[Ref. 1]

Catastrophic thermal failure is defined as an immediate, thermally induced, total loss of electronic function in a specified component. This type of failure is the result of a component melting due to excessive temperature, a thermal fracture of the substrate or carrier, or a separation of leads and the external network. It is generally considered to be dependent on the local temperature field, operating history, and operating modes of the component. As previously stated, a variety of problems arise when components are subjected to temperatures in excess of their rated limits. Furthermore, it is extremely difficult to determine the precise temperature at which catastrophic failure may occur. The incorporation of an accurate thermal analysis, in combination with test and operating experience, may be used to generate a catastrophe free upper operating limit. These maximum allowable operating temperatures are used to generate the master thermal control configuration for the system.[Ref.1]

#### III. HEAT TRANSFER

Heat transfer is defined as all energy flows that arise as a result of temperature differences [Ref. 6]. Because electronic components are not one hundred percent efficient, they produce heat as well as the desired output. In the case of semiconductor devices, heat develops in parts having low thermal efficiencies, such as the die. One of the major objectives of packaging is to develop an effective system for the removal of heat from these parts [Ref. 4]. It is imperative that design engineers understand all modes of heat transfer in order to incorporate an efficient method of heat removal into component designs. The modes of heat transfer are conduction, convection, and radiation.

#### A. CONDUCTION

Conduction is the transfer by molecular motion of heat between one part of a body to another part of the same body or between one body and another in physical contact [Ref. 1]. Joseph Fourier, a French physicist, proposed that the rate of heat flow through a material by conduction is proportional to the area of the material normal to the heat flow path and to the temperature gradient along the heat flow path.

This proportionality is represented mathematically by

$$q \propto -A \frac{dT}{dx} \tag{5}$$

where the minus sign allows for a positive heat flow in the presence of a negative temperature gradient. The introduction of a proportionality constant, known as thermal conductivity, results in the following rate equation which describes this mechanism [Ref. 1]:

$$q = -kA \frac{dT}{dx} \tag{6}$$

where

k = thermal conductivity of the material, W/m-C A = area of the heat flow path,  $m^2$  dT/dx = change in temperature per unit length, C/m q = rate of heat flow, W

#### 1. General Equation of Heat Conduction

The first step in the analytical solution of a heat conduction problem for a given structure is to choose an orthogonal coordinate system such that the surfaces coincide with the boundary surfaces of the structure [Ref. 7]. In the case of the model builder developed in this thesis, the

rectangular coordinate system will be employed. The general equation of heat conduction is given as

$$\frac{\partial}{\partial x} \left( k \frac{\partial T}{\partial x} \right) + \frac{\partial}{\partial y} \left( k \frac{\partial T}{\partial y} \right) + \frac{\partial}{\partial z} \left( k \frac{\partial T}{\partial z} \right) + q_{z} = \rho C \frac{\partial T}{\partial t}$$
 (7)

Then, assuming k, C, and  $\rho$  are independent of temperature, direction, and time, the resulting equation is

$$\frac{\partial^2 T}{\partial x^2} + \frac{\partial^2 T}{\partial y^2} + \frac{\partial^2 T}{\partial z^2} + \frac{q}{k} = \frac{1}{\alpha} \frac{\partial T}{\partial t}$$
 (8)

where

T = temperature, C

x, y, and z = cartesian coordinates, m

t = time, sec

k = thermal conductivity, W/m-C

q = internal heat generation, W/m<sup>3</sup>

 $\alpha$  = thermal diffusivity, k/ $\rho$ C, m<sup>2</sup>/sec

There are several variations of the general equation of conduction. The first, known as the Fourier equation, provides a solution for a system that contains no heat sources:

$$\frac{\partial^2 T}{\partial x^2} + \frac{\partial^2 T}{\partial y^2} + \frac{\partial^2 T}{\partial z^2} = \frac{1}{\alpha} \frac{\partial T}{\partial t}$$
 (9)

The second variation, known as the Poisson equation, supplies a solution for a system in which the temperature does not vary with time:

$$\frac{\partial^2 T}{\partial x^2} + \frac{\partial^2 T}{\partial y^2} + \frac{\partial^2 T}{\partial z^2} + \frac{q}{k} = 0$$
 (10)

The third and final variation of the general equation of conduction provides a solution for a system void of heat sources and operating in steady state. The resulting equation, known as the Laplace equation, is given as

$$\frac{\partial^2 T}{\partial x^2} + \frac{\partial^2 T}{\partial y^2} + \frac{\partial^2 T}{\partial z^2} = \nabla^2 T = 0$$
 (11)

#### 2. Simple Plane Slab

Consider Figure 7 which illustrates a simple plane slab with face temperatures  $T_1$  and  $T_2$ . Using only one dimension, equation (11) is reduced to

$$\frac{d^2T}{dx^2}=0$$

By integrating twice and applying boundary conditions the temperature distribution across the slab is seen to be

$$T = T_1 - \frac{X}{I} (T_1 - T_2)$$
 (12)

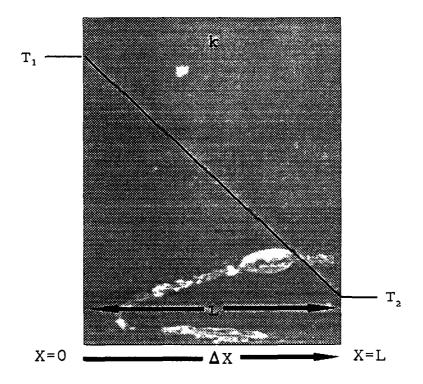


Figure 7. Conduction heat transfer through a simple plane slab.

Insertion of equation (12) into equation (6) produces a solution for the heat flow across the slab:

$$q = -kA \left[ \frac{-(T_1 - T_2)}{L} \right] = \frac{kA}{L} (T_1 - T_2)$$
 (13)

Ohm's Law indicates a direct analogy between heat flow, equation (13), and the flow of electrical current through a resistor, V = RI. This electrothermal analogy is extremely useful in the solution of one dimensional, steady state problems without energy generation and will be developed further in the next section. [Ref. 8]

#### 3. Electrothermal Analog

As previously stated, there is a direct analogy between heat flow across a simple plane slab, equation (13), and electrical current governed by Ohm's law:

$$I = \frac{V}{R} \tag{14}$$

In this case, the analogous quantities are

Current I → Heat Flow q

Potential V  $\rightarrow$  Temperature Difference  $\Delta T$ 

Resistance R + Thermal Resistance R

It is easily seen that for the heat flow in a simple plane slab described by, equation (13), the thermal resistance is

$$R = \frac{\Delta T}{Q} = \frac{L}{kA} \tag{15}$$

The electrothermal analog for conduction across a simple plane slab is shown in Figure 8.[Ref. 1]

#### B. CONVECTION

Convection is defined as the process by which thermal energy is transferred to or from a solid by a fluid flowing past it. Should the fluid flow be the result of a temperature difference the phenomena is called natural or free convection. On the other hand, when a pump or fan causes the mass movement the process is called forced convection. [Ref. 1]

Recall that at the interface between a solid and a fluid that heat is transferred by conduction and must obey Fourier's law, equation (6). Due to the difficulty encountered in accurately measuring the temperature gradient, Newton suggested that the surface heat transfer rate be related to the product of surface area and the temperature difference between the surface and the fluid. The results of this proposition lead to Newton's law of cooling:

$$q = hA(T_0 - T_f) \tag{16}$$

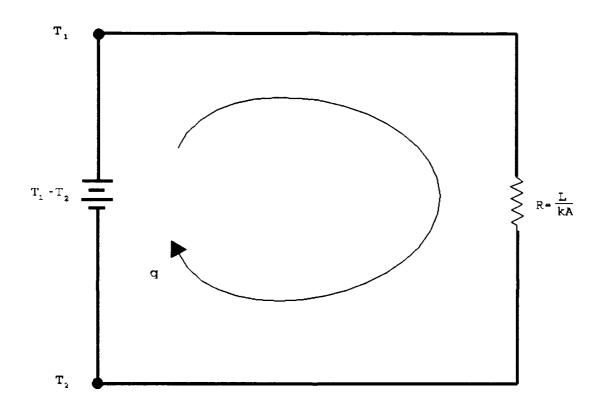


Figure 8. Conduction Electrothermal Analog.

where h is a proportionality factor that has become known as the surface heat transfer coefficient. [Ref. 1]

From a comparison of Newton's law of cooling, equation (16), and Fourier's law, equation (6), one can derive that the surface heat transfer coefficient can be related to the thermal conductivity, the wall temperature gradient of the fluid, and the surface fluid temperature difference:

$$h = \frac{q}{A \cdot \Delta T} = \frac{-k \left(\frac{\partial T}{\partial y}\right)}{\Delta T}$$
 (17)

Therefore, any correlation between heat transfer coefficients must reflect the dependence of h on the thermal conductivity of the fluid and on the ratio of the wall temperature gradient to the temperature difference. [Ref. 1]

#### 1. Electrothermal Analog

The addition of heat transfer by convection to both surfaces of the simple plane slab of Figure 7 results in the configuration shown in Figure 9. In the convective case thermal resistance is represented by

$$R = \frac{1}{hA} \tag{18}$$

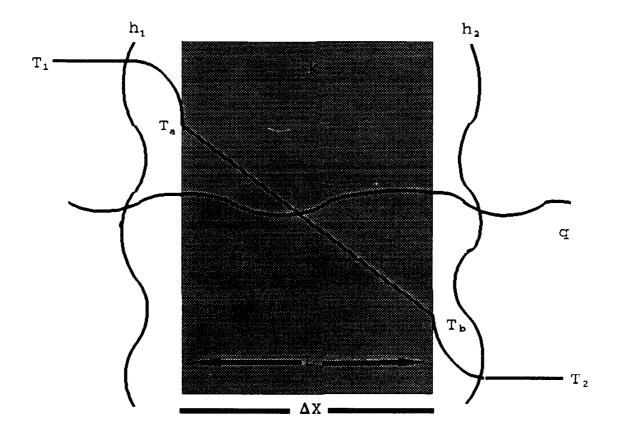


Figure 9. Convective heat transfer on a simple plane slab.

Therefore, the total thermal resistance is

$$R = \frac{1}{h_1 A} + \frac{L}{k A} + \frac{1}{h_2 A} = \frac{1}{A} \left[ \frac{1}{h_1} + \frac{L}{k} + \frac{1}{h_2} \right]$$
 (19)

which is represented by the electrothermal analog shown in Figure 10. A simple consideration of circuit theory then shows that:

$$q = \frac{\Delta T}{R} = \frac{(T_1 - T_2)}{\frac{1}{A} \left[ \frac{1}{h_1} + \frac{L}{k} + \frac{1}{h_2} \right]}$$
 (20)

#### C. RADIATION

Heat transfer by radiation is the means by which thermal energy can be transmitted through a space without an intervening medium while obeying the laws of electromagnetics. Thermal radiation, while traveling at the speed of light, may be absorbed, reflected, or transmitted upon contact with a surface. An ideal black body absorbs all incident radiation and reflects and transmits none of it. The concept of the black body is useful because laws governing its radiation are simple and many real bodies may be treated approximately as black bodies [Ref. 1].

Materials used in the fabrication of electronic components are classified as gray. Gray bodies are diffusely reflecting

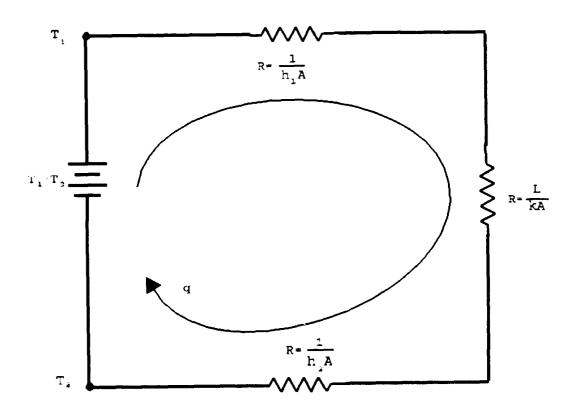


Figure 10. Conduction and convection electrothermal equivalent.

opaque surfaces [Ref. 4]. These surfaces reflect equal amounts of energy over the thermal radiation spectrum (wavelengths of about 0.1  $\mu$ m to about 100  $\mu$ m) in all directions. The heat transfer efficiency of this mode depends on the configuration, the orientation, and the temperatures of the surfaces in the electronic assembly.

#### 1. Transformation of the General Radiation Equation

The use of the thermal radiation equation in analytical studies is made difficult by its dependence on the fourth power relationship between the temperatures. Due to the nonlinear characteristics and the complexity of the calculations a computer program is the desired method to solve problems that have a significant transfer by radiation. The general equation for radiation interchange is:

$$q = \sigma F_a F_c A \left( T_s^A - T_r^A \right) \tag{21}$$

where

 $\sigma = \text{Stefan-Boltzman constant}, 5.669 \cdot 10^{-8} \text{ W/m}^2 - \text{ K}^4$ 

 ${\bf F}_{\rm A}={\bf s}$  hape factor that accounts for the arrangement of the of the radiating source and absorbing receiver

F<sub>e</sub> = emissivity factor that accounts for the
ability of the source and receiver to emit or absorb
radiation

 $T_s$  = temperature of the source, K

 $T_r = temperature of the receiver, K$ 

 $A = surface area, m^2$ 

It is important to note that the absolute temperature scale must be used when considering radiation.

One way to handle computations involving heat transfer by radiation is to transform the general radiation equation, equation (21), into a configuration compatible with Fourier's law. In this case, linearization of the general radiation equation is the method used to produce the desired result. This is achieved by factoring the difference in the fourth power of the temperatures as follows:

$$(T_s^A - T_r^2) = (T_s^2 + T_r^2) (T_s^2 - T_r^2)$$

$$= (T_s^2 + T_r^2) (T_s + T_r) (T_s - T_r)$$
(22)

Inserting this into equation (21) results in

$$q = \sigma F_A F_e A \left( T_S^2 + T_I^2 \right) \left( T_S + T_I \right) \left( T_S - T_I \right)$$
 (23)

A radiative heat transfer coefficient may therefore be defined as

$$h_r = \sigma F_a F_F (T_S^2 + T_R^2) (T_S + T_R)$$
 (24a)

or

$$h_{r} = \sigma F_{3} F_{E} (T_{S}^{3} + T_{S}^{2} T_{E} + T_{S} T_{R}^{2} + T_{R}^{3})$$
 (24b)

Then, substituting  $h_r$  into equation (23), radiation heat transfer may be treated exactly as convection at the boundary.

A thermal resistance for radiation heat transfer can now be proposed:

$$R = \frac{1}{h_r A} \tag{25}$$

Figure 11 provides an illustration of the electrothermal equivalent with the addition of radiation resistance in parallel with convective resistance.

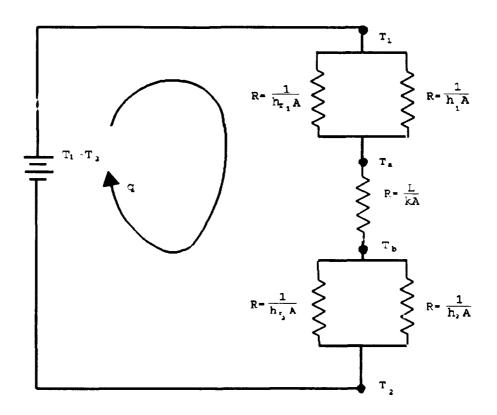


Figure 11. Radiation, convection, and conduction electrothermal equivalent.

### 2. General Problems of Heat Transfer by Radiation

In the calculation of heat transfer by radiation, it is usually necessary to approximate real-body behavior by the gray-body idealization. The assumption that emissivity is always equal to absorptivity is frequently required if the problem is to be solved. Even a simple situation becomes quite complex if real body behavior is considered and the resulting convenience of replacing absorptivity with emissivity is lost. A second difficulty in considering real-body behavior is the lack of sufficient data. To properly account for real-body behavior, extensive tabulations would be required. [Ref. 6]

The usual problem of heat transfer by radiation is further complicated not only because the surfaces are nonblack but also because the configuration of the areas involved is not simple and reflecting surfaces may be present to augment the direct exchange [Ref. 6]. There are a number of methods available to assist in the solution of heat transfer by radiation, however, their development is beyond the scope of this thesis.

### IV. FINITE DIFFERENCES

need to utilize a computer to determine temperature distribution within an electronic component or system has made finite difference methods very desirable. This useful numerical method is in problems involving nonlinearities, complex geometries, complicated boundary conditions, or a system of coupled partial differential equations. The purpose of this section is to provide the reader with some basic concepts involved in finite difference methods for solving differential equations. Furthermore, it demonstrates the methodology used to formulate n-node equations in *n*-unknown temperatures and instills confidence that this numerical method is capable of generating an accurate thermal model.

# A. FUNDAMENTAL CONCEPTS

Consider equation (7), the general equation for heat transfer by conduction. In order to produce a numerical solution to a conduction heat transfer problem it is necessary to reconfigure the partial differential equation into a form that allows differentiation to be performed by numerical methods. Therefore, it is essential that accurate approximations of the first and second derivatives be obtained.

## 1. First Derivative Approximation

The derivative of a function at any point can be expressed by a finite difference approximation by incorporating a Taylor series expansion about that point. Consider Figure 12 where T(x) is a function that can be expanded by a Taylor series. The Taylor series expansions of the functions  $T(x+\Delta x)$  and  $T(x-\Delta x)$  about a point x are:

$$T(x+\Delta x) = T(x) + \Delta x T'(x) + \frac{(\Delta x)^2}{2!} T''(x) + \frac{(\Delta x)^3}{3!} T'''(x) + \dots$$
 (26a)

$$T(x-\Delta x) = T(x) - \Delta x T'(x) + \frac{(\Delta x)^2}{2!} T''(x) - \frac{(\Delta x)^3}{3!} T'''(x) + \dots$$
 (26b)

In order to determine the first derivative, equations (26a) and (26b) are solved for T'(x).

$$T'(x) = \frac{T(x + \Delta x) - T(x)}{\Delta x} - \frac{(\Delta x)}{2} T''(x) - \frac{(\Delta x)^2}{6} T'''(x) - \dots$$
 (27)

$$T'(x) = \frac{T(x) - T(x - \Delta x)}{\Delta x} + \frac{(\Delta x)}{2} T''(x) - \frac{(\Delta x)^2}{6} T'''(x) + \dots$$
 (28)

Subtracting equation (26b) from (26a) and solving for T'(x) produces

$$T'(x) = \frac{T(x + \Delta x) - T(x - \Delta x)}{2\Delta x} - \frac{(\Delta x)^2}{6} T'''(x) - \dots$$
 (29)

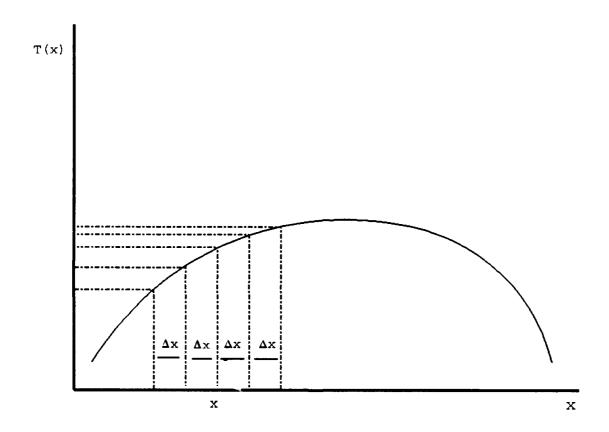


Figure 12. First and second derivative approximations.

From equations (27), (28), and (29) the first derivative approximations of T'(x) are determined to be

$$T'(x) = \frac{dT}{dx} = \frac{T(x + \Delta x) - T(x)}{\Delta x}$$
 forward difference (30)

$$T'(x) = \frac{dT}{dx} = \frac{T(x) - T(x - \Delta x)}{\Delta x}$$
 backward difference (31)

$$T'(x) = \frac{dT}{dx} = \frac{T(x + \Delta x) - T(x - \Delta x)}{2\Delta x} \quad central \ difference$$
 (32)

and for all it is observed that

$$T'(x) = \frac{dT}{dx} = \frac{\Delta T}{\Delta x}$$

# 2. Second Derivative Approximation

To obtain the second derivative of the function T(x), Figure 12, consider the Taylor series expansions of the functions  $T(x+2\Delta x)$  and  $T(x-2\Delta x)$  about a point x.

$$T(x+2\Delta x) = T(x) + 2\Delta x T'(x) + 2(\Delta x)^2 T''(x) + \frac{4}{3}(\Delta X)^3 T'''(x) + \dots$$
 (33a)

$$T(x-2\Delta x) = T(x) - 2\Delta x T'(x) + 2(\Delta x)^2 T''(x) - \frac{4}{3}(\Delta x)^3 T'''(x) + \dots$$
 (33b)

Inserting the corresponding first derivative approximation into equations (33a) and (33b) and solving for T''(x) results in

$$T''(x) = \frac{d^2T}{dx^2} = \frac{T(x) + T(x + 2\Delta x) - 2T(x + \Delta x)}{(\Delta x)^2} \quad \text{forward difference}$$
 (34)

$$T''(x) = \frac{d^2T}{dx^2} = \frac{T(x - 2\Delta x) + T(x) - 2T(x - \Delta x)}{(\Delta x)^2} \quad backward \ difference$$
 (35)

The central difference is obtained by eliminating T'(x) between equations (26a) and (26b).

$$T''(x) = \frac{d^2T}{dx^2} = \frac{T(x - \Delta x) + T(x + \Delta x) - 2T(x)}{(\Delta x)^2}$$
 central difference (36)

and for all it is observed that

$$T''(x) = \frac{d^2T}{dx^2} = \frac{(\Delta T)}{(\Delta x)^2}$$

# B. NODE ANALYSIS

The first step in the physical formulation of a solution by node analysis is, as previously stated, to divide the region into a finite number of subvolumes as shown in Figure 13. The centroid of each subvolume is called a node and is considered to be representative of the entire subvolume.

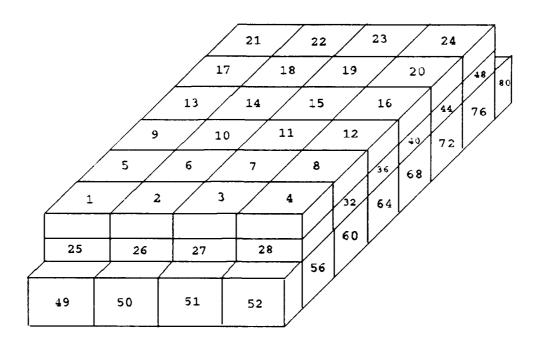


Figure 13. Structure divided into subvolumes.

At this point, continuity must be applied. In the steady state, the algebraic sum of all the heat leaving any node must equal zero. As an example, consider an interior node on the upper layer; select node number 14. Figure 14 provides an expanded picture around node-14. In order to maintain clarity the external environment will be considered as a single node, number 81. The resulting energy balance or node equation can be written as

$$q_{14,10} + q_{14,13} + q_{14,15} + q_{14,18} + q_{14,38} + q_{14,81} + q_{1} = 0$$
 (37)

Here, each numerical subscript represents the heat flow from node-14 to the indicated node. Furthermore,  $\mathbf{q}_i$  allows for energy input by an external source such as a dissipating component.

The energy balance simply supplies the pertinent energy terms. It is important to note that differences in thicknesses and thermal conductivities must be taken into consideration. The rate equations may be written in terms of thermal resistances and temperature differences:

$$q_{14,13} = \frac{k_1 \Delta x \Delta z_1}{\Delta v} (T_{14} - T_{13})$$
 (38a)

$$q_{14,15} = \frac{k_1 \Delta x \Delta z_1}{\Delta y} (T_{14} - T_{15})$$
 (38b)

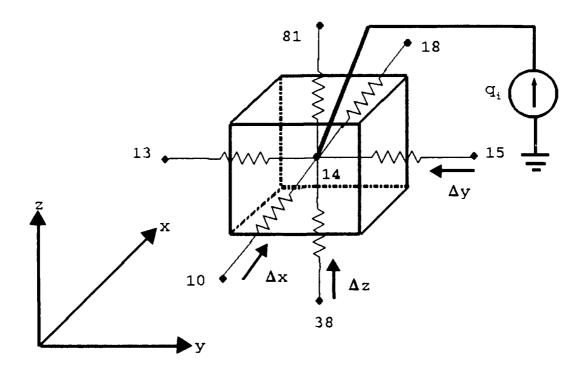


Figure 14.

$$q_{14,10} = \frac{k_1 \Delta y \Delta z_1}{\Delta x} (T_{14} - T_{10})$$
 (38c)

$$q_{14,18} = \frac{k_1 \Delta y \Delta z_1}{\Delta x} (T_{14} - T_{18})$$
 (38d)

$$q_{14,81} = \frac{2k_1 \Delta x \Delta y}{\Delta z_1} (T_{14} - T_{81})$$
 (38e)

$$q_{14,38} = \frac{2\Delta x \Delta y}{(\frac{\Delta z_1}{k_1}) + (\frac{\Delta z_2}{k_2})} (T_{14} - T_{38})$$
 (38f)

where

 $k_1$  and  $k_2$  = thermal conductivities of the top layer and middle layer, respectively, W/m- C

 $\Delta x$  and  $\Delta y$  = path lengths in their respective direction and are node dimensions for surface area calculations, m

 $\Delta z_1$  and  $\Delta z_2$  = the thicknesses for the upper and middle layers, respectively, m

 $T_{x}$  = temperature of the node indicated by the subscript,  $\cdot C$ 

Substituting these rate equations into the energy balance equation, equation (37), results in an equation containing five unknown temperatures.

For simplicity let  $\Delta x = \Delta y = \Delta z_1 = \Delta z_2 = 1$  and  $k_1 = k_2 = k$ . Then

$$k(T_{14} - T_{10}) + k(T_{14} - T_{13}) + k(T_{14} - T_{15}) + k(T_{14} - T_{18}) + 2k(T_{14} - T_{81}) + k(T_{14} - T_{38}) + q_i = 0$$
(39)

further reduction results in

$$-T_{10} - T_{13} - T_{15} - T_{18} - 7T_{14} - T_{38} = \frac{q_i}{k} + 2T_{81}$$
 (40)

An equation of this type can be written for every node in the region whose temperature is unknown. This results in a set of n-equations that relate the n-unknown temperatures. There are a variety of methods available to solve for the unknown temperatures. In large structures the most desirable method would be a thermal analysis program.

### V. THE MODEL BUILDER

As previously stated, the model builder currently being used by the thermal analysis software package requires an inordinate amount of time for data entry. This thesis presents a specific model builder, TMDL, that will become part of the software package. TMDL provides a user friendly, rapid method to assist in the development of a thermal model with a specific geometry as in Figure 2. Specifically, it generates a properly formatted output data file for use by the thermal analyzer enroute to preparing an accurate listing of the temperature distribution of the structure.

# A. THE THERMAL ANALYZER INPUT DATA FILE

The model builder generates a data file from physical characteristics of the structure provided by the user. This data file, also called the thermal analyzer input data file, must be in a format that is completely acceptable to the thermal analyzer. Furthermore, it will consist of five lines and up to seven data sets. This section describes each line and data set and their relationship with TMDL. Figure 15 displays a partial data file.

Line one is the title line. It may be left blank or may contain up to 79 alphanumeric characters. The user selected title appears at the top of the data file.

	EXAMP	LE OUTPUT	DATA FI	LE				
740	6	0	0	0	0	0	0	2
0	0	0					_	_
750	50	6	2	4	6	0	0	0
0								
.05000		66700				78.00000		00 75 6000
		6.90000000				75:		2411
	551	21		521	111 43.855	2.047	LΙ	1.456
73.70	-	36.850	87.70	521	121	75	1 1	2421
6 36.85	11	31 36.850	87.70		43.855	2.047		1.456
7	21	41		521	131	75	11	2431
36.85		36.850	87.70		43.855	2.047		1.456
6	31	51	_	521	141	75	11	2441
36.85		36.850	87.70		43.855	2.047		1.456
6	41	61		521	151	75	11	2451
36.85		36.850	87.70		43.855	2.047		1.456
6	51	71	7	521	161	75	11	2461
36.85	0	36.850	87.70	9	43.855	2.047		1.456
7	61	81	7	521	171	75	11	2471
36.85	0	36.850	87.70	9	43.855	2.047		1.456
6	71	91	7	521	181	75	11	2481
36.85	0	36.850	87.70	9	43.855	2.047		1.456
6	81	101	7	521	191	75	11	2491
36.85	0	36.850	87.70	9	43.855	2.047		1.456
7	91	7541	7	521	201	75		2501
36.85	0	73.700	87.70	9	43.855	2.047		1.456
	551 .	121		11	211	75		2511
73.70	-	36.850	43.85	-	43.855	2.047		1.456
-	111	131		21	221		11	2521
36.85		36.850	43.85		43.855	2.047		1.456
	121	141		31	231		11	2531
36.85	_	36.850	43.85		43.855	2.047		1.456
6	131	151		41	241	2.047	11	2541 1.456
36.85		36.850	43.85		43.855 251		11	2551
-	141	161	42.06	51	43.855	2.047		1.456
36.85 6	151	36.850 171	43.85	61	261		11	2561
36.85		36.850	43.85		43.855	2.047		1.456
6	161	181	43.0	71	271		11	2571
36.85		36.850	43.85		43.855	2.047		1.456
6	171	191	43.0	81	281		511	2581
36.85		36.850	43.89		43.855	2.047		1.456
7	181	201	45.0.	91	291		11	2591
36.85		36.850	43.89		43.855	2.047		1.456
6	191	7541		101	301	_	511	2601
36.85		73.700	43.89		43.855	2.047	7	1.456
	551	221	<del>-</del> -	111	311	75	511	2611
73.70		36.850	43.8	55	43.855	2.047	7	1.456
6	211	231		121	321	75	511	2621
36.85	50	36.850	43.8	55	43.855	2.047	7	1.456

Figure 15. Output data file.

Line two is the problem data line. It has nine entries of which two are under user control, number of nodes under consideration and unit type. One entry, the number of constant temperatures is preset at six for this specific model. The remaining entries have applications to models associated with heaters, unique exponents, secondary heat input, temperature coefficients and curves, and nodes controlling fast heat. These entries are not applicable to this model and are preset to zero.

Line three places a zero at three points and is beyond the user's control. Therefore, no further discussion is required.

Line four is the problem capability line. This line defines the maximum values for the entries in line two. The first entry is 750, the number of nodes for which the analysis is dimensioned. This is significant because the first constant temperature will be assigned the node number 751. The second entry is 50 which is the maximum number of constant temperatures in accordance with the analyzer dimension statement. Therefore, it is possible to have 50 constant temperatures allocated from node 751 to 800. The third entry is preset to six for an application concerning heaters and is not applicable to this model. The balance of the entries in line four represent a listing of data sets that are required for the particular analysis at hand. TMDL uses three data sets that will be discussed in what follows.

Line five contains five items that concern the level of accuracy that the thermal analyzer will achieve. These entries are preset. Item one provides the desired level of accuracy between iterations. Because the thermal analyzer solves iteratively, there must be an error criterion at which point calculations cease. A number that is too small will cause the computer to run excessively and a number that is too large will not give the desired accuracy. Therefore, a trade off is necessary and such a tradeoff has indicated that 0.05 is satisfactory. Item two is a damping factor that is used between iterations to prevent temperature oscillations between iterations. The third entry is the maximum number of iterations. This prevents excessive computer time in the event of faulty input data. Item four is a convergence factor that adjusts the damping to close to the critical value. This means that once the computer determines convergence is occurring, slow convergence is increased to reduce computer time. The fifth and final entry in line five is the initial temperature at which the iterative process begins. It is input by the user.

Input data set one contains temperature dependent coefficients and is not used in this model.

Input data set two contains up to 50 constant temperature inputs. This model has six constant ambient temperatures input by the user.

Input data set three involves heaters for fast warm up and is not used in this model.

Input data set four contains all pertinent information concerning the *n*-node equations. Each node requires two lines of data. The odd numbered lines are used for specifying the nodes that interact with the node in question and the modes by which this interaction takes place. Consider Figure 14, line one of data set four is as follows;

### 6 7551 21 7521 111 7511 2411

This line of data is for node number one of aspect ratio selection one. The first entry provides the number of connections to that node. Entry two says node 755, an ambient temperature, is connected to the node in question and the one indicates that the connection is by conduction. The rest of the entries are read similarly with numbers between 7511 and 7561 indicating a conductive connection with the external environment. External heat input is defined by entry 9991.

The even lines are used for specifying the inter node conductance values of the corresponding entry in the previous line.

Input data set five is used only if there are unique exponents. Therefore, no further discussion is required.

Input data set six contains the initial temperature guesses corresponding to the number of nodes receiving

secondary heat and is not used in this model because they have been set by the last entry in line five.

Input data set seven indicates the number of comperature dependent heat input curves and is not used in this model.

### B. FEATURES

TMDL is presented to the user in discrete sections. Upon entry into the first section, the user is offered an optional overview. This option should be accepted on at least the first run. At the completion of the overview the user is prompted for a data file name and title. The data file name should be changed for each successive run because TMDL does not over write existing files.

The second section provides prompts for the physical characteristics of the structure. Initially, the user is given a list of four aspect ratios from which to choose;

- 1.) 10 by 24 nodes provides a 1:2.4 ratio with 720 nodes.
- 2.) 15 by 15 nodes provides a 1:1 ratio with 675 nodes.
- 3.) 12 by 20 nodes provides a 1:1.6 ratio with 720 nodes.
- 4.) 8 by 30 nodes provides a 1:3.75 ratio with 720 nodes.

It should be noted that up to 30 additional nodes may be added depending on the existence of the mounting ear. Figure 16 displays the specific geometry with definitions.

After selecting the desired aspect ratio, the user must specify the use of either SI or English units. The input of

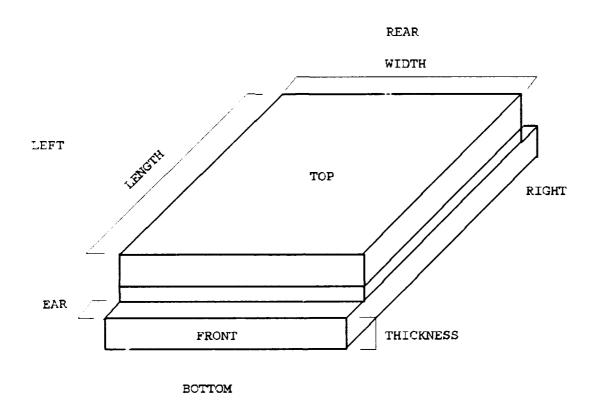


Figure 16.

data must remain consistent with this selection. At this point the input of structure characteristics are requested. These entries consist of the length, width, thickness, and thermal conductivity of each layer. There is ample opportunity for correction or alteration throughout TMDL.

Section three requests the input of initial and ambient temperatures. TMDL has six ambient temperatures.

The final section allows for the input of external heat sources. TMDL provides for heat input into the upper substrate surface only. There are four methods of external heat input from which to choose. The first alternative allows for a total rate of heat applied to the upper surface. An entry for this alternative would be divided by the nodes and distributed appropriately. Option two provides for the entry of average heat per unit area. The third alternative gives the user the freedom to enter heat in specifically selected nodes.

Option four offers no heat input.

At the conclusion of these entries an output data file will be created in the proper format and placed in a file named by the user.

### APPENDIX A

#### TMDL LISTING

```
$LARGE
 TITLE:
C
          MODEL BUILDER
C AUTHOR: LT PATRIC ROESCH
C DATE:
          15 MARCH 1991
C COMPILER: MICROSOFT VERSION 4.01
C LINKER: MICROSOFT VERSION 3.55
C DEFINE REAL VARIABLES
   REAL SL.SW, DELS, KS, EW, EL, DELF, KE, CL, CW, DELC, KC, IT, UPRT, LWRT, LT, RT
   +.FT,BT,THEAT,THPN,AHEAT,NHEAT,DELX,DELY,SYLR,SXFB,SXX,SYY,SZE,SZT
   REAL EXFB,EYY,EXX,FZC,EAR,CYLRE,CYYE.CXFBE,CXXE,CZEAR,CYLR,CYY,CXX
   +,CZB,ACC,DAMP,CONFAC,EYLR
C
C
C DEFINE INTEGER VARIABLES
   INTEGER NPL, NWIDE, NDEEP, NUM, NUMA, CH, H, TOTNOD, NN, NC, N, IB, IE, IC, ID,
   +COUNT.CONTEMP.USEL,ZER,NMAX,TMAX,HTRS,D1,D2,D3,D5,D4,D6,D7,MAXIT
C
C. DEFINE ALL ONE CHARACTER VARIABLES
   CHARACTER*1 ANS,ANSN,ANSA,ANSSL,ANSSW,ANSST.ANSSK,ANSE,ANSEL,
   +ANSEW,ANSET,ANSEK,ANSC,ANSCL,ANSCW,ANSCT,ANSCK,ANST,ANSTI,ANSTU,
   +ANSTLR,ANSTL,ANSTR,ANSTF,ANSTB,ANSH,ATH,ANSHA,AHN,SELECT,SELH
   +.ANSWER.DELT
C
C. CHARACTER VARIABLES OF MORE THAN ONE POSITION
   CHARACTER DATAF*70,UK*11,UT*1,UH*13,UAH*6,NAME*6,UL*2
C DEFINE MATRICES
   REAL HEAT(30,15),COEF(740,9)
C
   INTEGER IH(240), JH(240), NCON(740,9)
C
\mathbf{C}
C
  PROVIDE THE USER WITH A PROGRAM OVERVIEW
\mathbf{C}
1006 CALL CLS
   WRITE(*,1001)
1001 FORMAT(///.
       THIS PROGRAM WAS WRITTEN TO INTEGRATE WITH EXISTING 1/2.
       THERMAL ANALYSIS SOFTWARE AND TO REDUCE THE AMOUNT
       OF TIME REQUIRED FOR DATA ENTRY. //.
       WOULD YOU LIKE AN OVERVIEW OF THE PROGRAM PRIOR TO - 1/4
       BEGINNING? ENTER Y FOR YES OR N FOR NO. (2X.)
   READ(*.1002) OVR
1002 FORMAT(A1)
```

```
C
   IF (OVR.EQ.'Y'.OR.OVR.EQ.'v') THEN
    CALL CLS
     WRITE(*,1003)
1003
      FORMAT(#:
                        THIS PROGRAM PERFORMS A NODAL ANALYSIS OF A THREE
       LAYERED CONDUCTIVE STRUCTURE (SUBSTRATE, EPOXY, AND
       CARRIER). THE OUTPUT CONSISTS OF UP TO 740 COEFFI-
       CIENTS THAT CONTRIBUTE TO THE DETERMINATION OF
       THE TEMPERATURE DISTRIBUTION OF A MICROCIRCUIT
       WHEN FED INTO THE THERMAL ANALYZER. 12.
       THE FOLLOWING IS AN ORDERED OUTLINE OF THE MAJOR
       SECTIONS OF THIS PROGRAM AND WHAT ENTRIES ARE
       REQUIRED OF THE USER. 17,
       NOTE THAT ALL ENTRIES WILL BE IN UPPER CASE LETTERS. V.
       A.) DATA FILE ',,
         1.) ENTER THE TITLE OF THE OUTPUT DATA FILE. 1/2.
         2.) ENTER THE DATA FILE NAME. THIS PROGRAM WILL NOT: J.
           ERASE OR WRITE OVER EXISTING FILES. ',//,
      PLEASE PRESS ENTER WHEN READY TO CONTINUE. ',2X,\)
     READ(*,1002)ANSWER
     CALL CLS
     WRITE(*,1004)
1004
      FORMAT(///,
       B.) STRUCTURE PHYSICAL CHARACTERISTICS
         1.) SELECT UNIT TYPE (SI OR ENGLISH) ',
         2.) SELECT FROM FOUR ALTERNATIVES THE DESIRED NODAL',
           ASPECT RATIO. ',/,
         3.) FNTER EACH LAYER CHARACTERISTICS. SUBSTRATE IS ',/,
           THE TOP LAYER, FOLLOWED BY EPOXY AND THE CARRIER 1/1
           LAYERS. RESPECTIVELY. THE CARRIER LAYER MAY
           CONTAIN AN EAR ON THE FRONT AND BACK SURFACES W.
           IF SPECIFICATIONS REQUIRE IT.'//,
       PLEASE PRESS ENTER WHEN READY TO CONTINUE. 12X,0
    READ(*,1002)ANSWER
     CALL CLS
     WRITE(*,1005)
1005
     -FORMAT(iii,
      C.) INITIAL AND AMBIENT TEMPERATURES
         1.) ENTER INITIAL CHIP AND CHASIS TEMPERATURE. 7.
           ASSUME STEADY STATE FOR INITIAL TEMPERATURE. 1,
         2.) ENTER AMBIENT TEMPERATURES FROM ALL SIDES OF THE 1/2.
           STRUCTURE.',//,
      D.) HEAT INPUT ',
         HEAT INJECTION OCCURS ONLY ON THE UPPER SURFACE OF ',
         THE SUBSTRATE LAYER. THIS PROGRAM SUPPLIES THE USER ://,
         FOR ALTERNATIVE METHODS FOR ENTERING HEAT; 1/1,
         1.) TOTAL HEAT OVER SUFACE './.
         2.) AVERAGE HEAT PER UNIT AREA :/,
         3.) INPUT HEAT NODE BY NODE 1/4
         4.) NO HEAT INPUT ',//,
      THIS COMPLETES THE PROGRAM OVERVIEW. PLEASE PRESS ENTER ',/,
      TO CONTINUE. (,2X,\)
```

```
READ(*,1002)ANSWER
C
   ELSEIF (OVR.EQ.'N'.OR.OVR.EQ.'n') THEN
     GOTO 1007
   ELSE
     GOTO 1006
1007 ENDIF
C
C
\mathbf{C}
   INITIALIZE MATRICES
   DATA HEAT /450*0.0/
   DATA IH /240*0/
   DATA JH /240*0/
(
C
   VARIABLE, CONSTANT, AND STRING DEFINITION
(
     PHYSICAL CHARACTERISTICS
       SL,EL,CL - SUBSTRATE, EPOXY, AND CARRIER LENGTHS
(
       SW,EW,CW - SUBSTRATE,EPOXY. AND CARRIER WIDTHS
\mathbf{C}
\mathbf{C}
       KS,KE,KC - SUBSRATE,EPOXY,AND CARRIER THERMAL CONDUCTIVITIES
       DELS, DELC - SUBSTRATE, EPOXY AND CARRIER THICKNESS
C
       DELX - SL/NDEEP
       DELY - SW/NWIDE
(
C
       NPL - NUMBER OF NODES PER A LAYER
C
       NWIDE - NUMBER OF NODES WIDE
C
       NDEEP - NUMBER OF NODES DEEP
C
       UL - UNITS OF LENGTH (SI OR ENGLISH)
C
       UK - UNITS OF THERMAL CONDUCTIVITY (SI OR ENGLISH)
\mathbf{C}
       EAR - DEPTH OF THE EAR (SL-CL)/2
C
Ċ
      INITIAL AND AMBIENT TEMPERATURES
C
       IT - INITIAL CHASIS AND CHIP TEMPERATURE
C
       LT - LEFT SIDE AMBIENT TEMPERATURE
C
       RT - RIGHT SIDE AMBIENT TEMPERATURE
C
       FT - FRONT AMBIENT TEMPERATURE
(
       BT - BACK AMBIENT TEMPERATURE
       UPRT - UPPER AMBIENT TEMPERATURE
(
       LWRT - LOWER AMBIENT TEMPERATURE
(
       UT - UNITS OF TEMPERATURE (CENTIGRADE OR FARENHIET)
(
\mathbf{C}
(
      HEAT INPUT
       THEAT - TOTAL INJECTED HEAT
(
       THPN - TOTAL HEAT PER NODE
(
       AHEAT - AVERAGE HEAT OVER A GIVEN SURFACE
(
       NHEAT - HEAT PER NODE INJECTED NODE BY NODE
       UH AND UAH - UNITS OF HEAT (SI OR ENGLISH)
(
       NUM, NUMA, CH, H - DUMMY VARIABLES USED TO CREATE THE VECTORS
C
                 IH, & JH WHICH ARE USED TO RELATE NODE
C
                 NUMBER TO MATRIX POSITION
(
C
       TOTNOD, NN.NC, N - VARIABLES USED TO ALLOW HEAT INPUT NODALLY
\mathbf{C}
       IH.JH - VECTORS USED TO CORRELATE NODE NUMBER WITH
(
            MATRIX POSITION
\mathbf{C}
       HEAT - MATRIX USED TO CONTAIN HEAT INPUTS
```

C	COEFFICIENT DEFINITIONS Y IMPLIES WIDTH & X IMPLIES DEPTH
C	Z IMPLIES HIEGHT
(,	S IMPLIES SUBSTRATE, E IMPLIES EPOXY
C	C IMPLIES CARRIER
(,	
C	SYLR - PROVIDES COEFFICIENT FOR LEFT OR RIGHT EDGE NODES
C	TO THE EXTERNAL NODE. SYLR IMPLIES SUBSTRATE,
C	EYLR - EPOXY
Ċ	CYLR - CARRIER/ NO EAR
C	CYLRE - CARRIER W/ EAR
C	CILKL - CARRIER W/ LAR
	CVED DROWING COFFECIENT FOR FRONT AND RACK FROE MORES
C	SXFB - PROVIDES COEFFICIENT FOR FRONT AND BACK EDGE NODES
C	TO THE EXTERNAL NODE.
C	EXFB - EPOXY
C	CXFB - CARRIER/ NO EAR
C	CXFBE - CARRIER W/ EAR
C	
C	SYY - PROVIDES INTERNAL COEFFICIENT IN THE 'Y' DIRECTION
$\mathbf{C}$	EYY - EPOXY
$\mathbf{C}$	CYY - CARRIER W/NO EAR
(	CYYE - CARRIER W/ EAR
$\mathbf{C}$	
C	SXX - PROVIDE INTERNAL COEFFICIENT IN THE 'X' DIRECTION
Ċ	EXX - EPOXY
Ċ	CXX - CARRIER W/NO EAR
Ċ	CXXE - CARRIER W/ FAR
Ċ	CAME - CAMMIN WE FAN
C	SZT - COEFFICIENT FOR SUBSTRATE TO UPPER EXTERNAL NODE
C	
	SZE - COEFFICIENT FOR SUBSTRATE TO EPOXY
C	EZC - COEFFICIENT FOR EPOXY TO CARRIER
C	CZEAR - COEFFICIENT FOR EAR TO LOWER EXTERNAL NODE
C	CZB - COEFFICIENT FOR CARRIER TO LOWER EXTERNAL NODE
C	
C	CHARACTER
(`	ALL ONE CHARACTER STRINGS EXCEPT 'UT' ARE SIMPLE
C	YES, NO, OR SELECTION ANSWERS AND DO NOT REQUIRE
C	EXPLANATION.
C	
C	DATA FILE
C	DATAF - TITLE FOR DATA FILE, LINE ONE
C	NAME - DATA FILE NAME
Č	IB,IE,ID,IC - COUNTERS FOR FILLING DATA FILE
Č	COUNT - TOTAL NUMBER OF NODES
Ċ	ACC - MINIMUM ERROR CRITERIA AT WHICH CALCULATIONS CEASE
C	DAMP - DAMPING FACTOR FOR PREVENTION OF TEMPERATURE
Ċ	OSCILLATIONS
C	CONFAC - CONVERGENCE FACTOR USED TO INCREASE DAMPING
C	CONTEMP - TMDL HAS 6 CONSTANT AMBIENT TEMPERATURES
C	USEL - TELLS ANALYZER WHAT TYPF OF UNITS ARE IN USE
C	ZER - DUMMY VARIABLE
(,	NMAX - MAXIMUM NODES PROGRAM IS CAPABLE OF EVALUATING
C	TMAX - MAXIMUM NUMBER OF CONSTANT TEMPERATURES PROGRAM
C	CAN ACCEPT
(,	HTRS - NOT APPLICABLE TO THIS PROGRAM - NO EXPLANATION
C	NECCESSARY

```
(
        D1-7 - DATA SETS IN USE
(
        MAXIT - MAXIMUM ITERATIONS COMPUTER WILL PERFORM TO
             ACHIEVE SOLUTION
0
(
C BEGIN PROGRAM
   CALL CLS
   WRITE(*,406)
406 FORMAT( ...
            PLEASE SELECT UPPER CASE LETTERS ***;,
             PRIOR TO BEGINNING.
   WRITE(*.201)
201 | format( /, '
         PRIOR TO ENTERING DATA INTO THIS PROGRAM ENSURE THAT J,
         YOU HAVE A DRAWING OF YOUR DESIGN AND ALL PERTINENT ',
         DATA.
                                     ·,//,
         PRESS ENTER WHEN READY TO CONTINUE.(2X.)
   READ(*,304)ANSWER
C. ALLOW USER TO NAME THE DATA FILE
   CALL CLS
   WRITE(*,701)
701 FORMAT(□□ THIS PROGRAM CREATES AN OUTPUT DATA FILE FOR ENTRY
   + INTO THE Z.
      EXISTING THERMAL ANALYZER, FURTHERMORE, THIS PROGRAM DOES J.
      NOT ERASE OR WRITE OVER THE EXISTING DATA FILE. THEREFORE ...
     THE USER WILL NAME THE DATA FILE FOR EACH RUN OF THIS 11.
   +* PROGRAM. THE FILE NAME IS LIMITED TO SIX CHARACTERS. (2)
   +1 PLEASE ENTER THE DI SIRED DATA FILE NAME, 1,2X, )
   READ 702, NAME
702 FORMAT(A6)
(
(
  ALLOW USER TO PROVIDE TITLE LINE FOR DATA FILE
   CALL CLS
   WRITE(*,202)
202 FORMAT(////,
                  ENTER THE DESIRED TITLE TO BE PLACED ON LINE
         NUMBER ONE OF THE OUTPUT DATA FILE. ...
         `.2X,\)
   READ LDATAF
   FORMAT(A70)
1
(
   CALL CLS
  SUPPLY A LISTING OF ACCEPTABLE ASPECT RATIOS
```

```
(
   WRITE(*,203)
203 FORMAT(###,*
                   THE FOLLOWING ASPECT RATIOS ARE AVAILABLE.)
   WRITE(*,204)
204 FORMAT(/,*
               1.) 10 BY 24 PROVIDES A 1:2.4 RATIO WITH 740 NODES.
       2.) 15 BY 15 PROVIDES A 1:1 RATIO WITH 675 NODES.'./.
   +'...
      3.) 12 BY 20 PROVIDES A 1:1.6 RATIO WITH 720 NODES. /
        4.) 8 BY 30 PROVIDES A 1:3.75 RATIO WITH 720 NODES. at
        PLEASE SELECT A NUMBER ONE THROUGH FOUR. (2X.)
   READ(*.3020)SELECT
3020 FORMAT(A1)
\mathbf{C}
C. USER CAN CONTINUE OR MAKE ANOTHER SELECTION
   CALL CLS
   4F (SELECT.NF (CAND.SELECT.NE.2).AND.SELECT.NF.3).AND.SELECT.
   +NE.4') THEN
     GOTO 7
   ELSE
10
      WRITE(*,3) SELECT
     FORMAT( - . . .
                        YOU SELECTED NUMBER ',A1,' OF THE FOLLO
   +WING 4 ALTERNATIVES....
          1.) 10 BY 24 FOR A 1:2.4 RATIO (2.)
          2.) 15 BY 15 FOR A 1:1 RAΓΙΟ΄,
          3.) 12 BY 20 FOR A 1:1.6 RATIO ...
          4.) 8 BY 30 FOR A 1:3.75 RATIO (a)
(
   ENDIF
\mathbf{C}
   WRITE(*,205)
205 FORMATC
                   IS THIS THE DESIRED SELECTION. ENTER Y FOR YES AN
   +D N FOR NO. (2X<sub>0</sub>)
   READ(*, 8)ANS
   FORMAT(A1)
(
C
   IF (ANS.EQ.'N') THEN
     GOTO 7
   ELSEIF (ANS.EQ.'Y') THEN
     GOTO 9
   ELSE
     CALL CLS
     GOTO 10
u
   ENDIF
(
(
   CALL CLS
(
(
C. AFTER CONFIRMATION OF SELECTION FILL CONSTANTS WITH APPROPRIATE
C VALUES
   IF (SELECTEQ.T) THEN
     NWIDE = 10
     NDEEP = 24
     NPL = 240
(
```

```
ELSEIF (SELECT.EQ.2') THEN
     NWIDE = 15
     NDEEP = 15
     NPL = 225
   ELSEIF (SELECT.EQ.3') THEN
     NWIDE = 12
     NDEEP = 20
     NPL = 240
\mathbf{C}
   ELSEIF (SELECT.EQ.4') THEN
     NWIDE = 8
     NDEEP = 30
     NPL = 240
   ENDIF
\mathbf{C}
C MAKE DESIRED UNIT SELECTION
C
  WRITE(*,206)
206 FORMAT(#####.
                    THIS PROGRAM IS CAPABLE OF OPERATIONS IN EI
   +THER SLOR L.
        ENGLISH UNITS. AFTER THE SELECTION OF THE UNITS, ALL V.
        ENTRIES MUST BE COMPATIBLE. PLEASE MAKE YOUR SELECTION: /
         S FOR SUNOTATION ...
        E FOR ENGLISH NOTATION (2X...)
   READ(*,302)ANSN
302 FORMAT(A1)
(
C CHECK FOR CORRECT UNIT SELECTION
  IF (ANSN.EQ.'S') THEN
     WRITE(*,207)
      FORMAT(///.:
                     YOU SELECTED SI NOTATION.')
   ELSEIF (ANSN.EQ.'E') THEN
     WRITE(*.208)
                     YOU HAVE SELECTED ENGLISH NOTATION.')
208
      FORMAT(/////.`
   ELSE
     CALL CLS
     GOTO 2
   ENDIF
C
   WRITE(*.209)
209 FORMAT(
                IS THIS THE DESIRED SELECTION? ENTER Y FOR YES AND
   +N FOR NO. 1.2X.\()
   READ(*,303)ANSA
303 FORMAT(A1)
C. SHOULD I CONTINUE OR RESELECT UNITS
   IF (ANSA.EQ.'Y') THEN
     GOTO 6
   FISHF (ANSA.FQ.'N') THEN
     CALL CLS
     GOTO 2
   ELSE
     CALL CLS
```

```
WRITE(*,210)
    FORMAT(il/inii.
                  THIS PROGRAM IS CAPABLE OF OPERATIONS IN
210
  + EITHER SLOR'J.
       ENGLISH UNITS. AFTER THE SELECTION OF UNITS. ALL'4.
       ENTRIES MUST BE COMPATIBLE. PLEASE MAKE YOUR SELECTION...
       S FOR SUNOTATION...
       E FOR ENGLISH NOTATION.)
(
    GOTO 4
   ENDIF
()
\cdot
   CALL CLS
C
       ****** ENTER CHIP CHARACTERISTICS FOR EACH LAYER*********
C
C
      C^*
\mathbf{C}
   WRITE(*,211)
C PROVIDE CORRECT UNIT ABBREVIATIONS
   IF (ANSN.EQ.S.) THEN
    WRITE(*.212)
    FORMAT( ALL ENTRIES ARE IN SI NOTATION...)
    UL = 'cm'
    UK = Watts cm/C^{-1}
    UT = C
   ELSEIF (ANSN.EQ.E.) THEN
    WRITE(*.213)
    FORMAT( ALL ENTRIES ARE IN ENGLISH NOTATION. //)
213
    UL = in'
    UK = Btu/hr/in/F
    UT = T
   ENDIF
C
C
(,
   WRITE(*,220)UL
220 FORMAT('ENTER SUBSTRATE LENGTH (',A2,'):
                                          1.2X_{\cdot}
   READ *,SL
(
   WRITE(*,216)UL
216 FORMAT( .: ENTER SUBSTRATE WIDTH (',A2,'):
                                          (.2X.)
   READ *.SW
(
   WRITE(*.217)UL
217 FORMAT(:: ENTER SUBSTRATE THICKNESS (I.A2.):
                                          '.2X,.)
```

```
READ *, DELS
C
   WRITE(*,218)UK
218 FORMAT(/,' ENTER SUBSTRATE THERMAL CONDUCTIVITY (',A12,'): ',2X,\)
   READ *,KS
C
31 CALL CLS
\mathbf{C}
C. MAKE ANY CHANGES OR CORRECTIONS TO SUBSRATE ENTRIES
   WRITE(*,214)
214 FORMAT(##,
                   YOU HAVE MADE THE FOLLOWING ENTRIES FOR THE SUB
   +STRATE...)
20 PRINT *C
              L) LENGTH SL
              2.) WIDTH SW
   PRINT *:
   PRINT *,
              3.) THICKNESS IDELS
   PRINT *."
              4.) k
                      :,KS
(
   WRITE(*,215)
215 FORMAT(:
                  DO YOU WISH TO MAKE ANY CHANGES? SELECT Y FOR YES
   + AND N FOR NO. 1,2X,(i)
   READ(*,304)ANSS
304 FORMAT(A1)
\mathbf{C}
   IF (ANSS.EQ.Y') THEN
12
     CALL CLS
     WRITE(*.219)
219
     FORMAT(////)
     PRINT *; THE CURRENT ENTRY FOR LENGTH IS ',SL,' ',UL
     WRITE(*,2190)
     FORMAT(/,
                   WOULD YOU LIKE TO CHANGE THE LENGTH? (Y OR N)
   + (.2X.)
     READ(*.304)ANSSL
     PRINT *
       IF (ANSSL.EQ.'Y') THEN
         WRITE(*.221)UL
221
         FORMAT(<) ENTER SUBSTRATE LENGTH ('.A2.'): '.2X.()
        READ *.SL
       ELSEIF (ANSSLEQ.N') THEN
        GOTO 14
       ELSE
        GOTO 12
       ENDIF
(
14
     CALL CLS
     WRITE(*,2191)
2191
     FORMAT(IIII)
     PRINT *.' THE CURRENT ENTRY FOR WIDTH IS ',SW,' ',UL
     WRITE(*,222)
222
                   WOULD YOU LIKE TO CHANGE THE WIDTH? (Y OR N) ',
     FORMAT(;;)
   +2X.3
     READ(*,2192) ANSSW
2192
     FORMAT(A1)
       IF (ANSSW.EQ. Y') THEN
         WRITE(*.223)UL
```

```
223
         FORMAT( : ENTER SUBSTRATE WIDTH ('.A2.'): '.2X.\)
        READ *.SW
       ELSEIF (ANSSWIPQIN') THEN
        GOTO 13
       ELSE
        GOTO 14
       ENDIF
(
\mathbf{C}
13
     CALL CLS
     WRITE(*,2193)
2193
     FORMAT(////)
     PRINT *, THE CURRENT ENTRY FOR THICKNESS IS ',DELS,' ',UL
     WRITE(*,224)
224
      FORMAT(/,
                   WOULD YOU LIKE TO CHANGE THE THICKNESS? (Y OR N)
   + ',2X,\)
     READ(*.2194) ANSST
2194
     FORMAT(A1)
       IF (ANSST.EQ.'Y') THEN
         WRITE(*,225)UL
225
         FORMAT(:,
                     ENTER SUBSTRATE THICKNESS (',A2,'): '.2X,
   + )
        READ *,DELS
       ELSEIF (ANSST.EQ.'N') THEN
        GOTO 15
       ELSE
        GOTO 13
       ENDIF
\mathbf{C}
(
15
     CALL CLS
     WRITE(*,2195)
     FORMAT(##/)
2195
     PRINT *.' THE CURRENT ENTRY FOR THERMAL CONDUCTIVITY IS ',KS
   +.::.UK
     WRITE(*,226)
   FORMAT(/, WC+Y? (Y OR N) ',2X.\)
                 WOULD YOU LIKE TO CHANGE THE THERMAL CONDUCTIVIT
226
     READ(*,2196) ANSSK
2196
    FORMAT(A1)
       IF (ANSSK.EQ.'Y') THEN
         WRITE(*,227)UK
         FORMAT(.)
                     ENTER SUBSTRATE THERMAL CONDUCTIVITY (,A1
   +2.): (2X.
        READ *.KS
       ELSEIF (ANSSK.EQ.'N') THEN
        GOTO 16
       ELSE
        GOTO 15
       ENDIF
(
  ALLOW ANOTHER REVIEW OF SUBSTRATE DATA ENTRIES
(
16
     CALL CLS
     WRITE(*,228)
```

```
YOU HAVE MADE THE FOLLOWING CORRECTIONS TO T
   FORMAT(///,
  +HE SUBSTRATE ENTRIES.™)
    GOTO 20
C
   ELSEIF (ANSS.EQ.'N') THEN
    GOTO 17
   ELSE
    GOTO 31
   ENDIF
\mathbf{C}
C
C****
         \mathbf{C}
C
17
   CALL CLS
   WRITE(*,229)
229 FORMAT(///,'***********************
  +*******,/,
  +********//,
  +' NOTE THAT SUBSTRATE AND EPOXY LENGTH AND WIDTH ARE EQUAL. '\
(
   WRITE(*,230)UL
230 FORMAT(/,' ENTER FPOXY THICKNESS ('.A2.'): '.2X.\)
   READ *,DELE
   WRITE(*,231)UK
231 FORMAT(a) ENTER EPOXY THERMAL CONDUCTIVITY (LA12.): (2X3)
   READ * KE
(
   EL = SL
   EW = SW
(
C REVIEW EPOXY ENTRIES
(,
30 CALL CLS
   WRITE(*,232)
              YOU HAVE MADE THE FOLLOWING ENTRIES FOR THE EPOXY
232 FORMAT(///,
   + LAYER. (A)
32 PRINT *, 1.) LENGTH PRINT *, 2.) WIDTH
                              :.EL
                             `.EW
   PRINT *: 3.) THICKNESS
                               '.DELE
   PRINT * . 4.) k
                           .KF
(`
   WRITE(*,233)
              NOTE THAT CHANGING EPOXY LENGTH OR WIDTH ALSO CHANG
233 FORMAT(/,
   +ES'... SUBSTRATE LENGTH OR WIDTH. //.
   + DO YOU WISH TO MAKE ANY CHANGES TO THE EPOXY ENTRIES? (Y OR
   +N) (2X..)
   READ(*,2197) ANSE
219" FORMAT(A1)
```

```
C MAKE ANY CHANGES OR CORRECTIONS TO EPOXY ENTRIES
C
   IF (ANSE.EQ.'Y') THEN
22
      CALL CLS
     WRITE(*,2198)
2198
     FORMAT(#//)
     PRINT *,' THE CURRENT ENTRY FOR LENGTH IS ',EL.' ',UL
     WRITE(*,234)
234
     FORMAT( ;
                  WOULD YOU LIKE TO CHANGE THE LENGTH? (Y OR N)
   +1,2X,1)
     READ(*,2199) ANSFL
2199
      FORMAT(A1)
(
       IF (ANSELLEQ.Y') THEN
         WRITE(*,235)UL
235
         FORMAT(/, ENTER EPOXY LENGTH ('.A2.'): '.2X.\)
         READ *,EL
         SL = EL
       ELSEIF (ANSEL.EQ.'N') THEN
         GOTO 21
       ELSE
         GOTO 22
21
       ENDIF
(
24
     CALL CLS
     WRITE(*,661)
601
      FORMAT(////)
     PRINT *,' THE CURRENT ENTRY FOR WIDTH IS ',EW,' ',UL
     WRITE(*,236)
236
     FORMAT(/,
                   WOULD YOU LIKE TO CHANGE THE WIDTH? (Y OR N)
   +1.2X.
     READ(*.602)ANSEW
602
      FORMAT(A1)
C
       IF (ANSEW.EQ.Y') THEN
         WRITE(*,237)UL
237
         FORMAT( .:
                     ENTER EPOXY WIDTH ('.A2.'): '.2X.')
         READ * FW
        SW = EW
       ELSEIF (ANSEW.EQ.N') THEN
        GOTO 23
       ELSE
         GOTO 24
23
       ENDIF
(
26
     CALL CLS
     WRITE(*,603)
      FORMAT(##)
603
     PRINT *,* THE CURRENT ENTRY FOR THICKNESS IS ',DELE,' ',UL
     WRITE(*,238)
238
                   WOULD YOU LIKE TO CHANGE THE THICKNESS? (Y OR N)
     FORMAT(*,*
   + (.2X_{.})
     READ(*,604)ANSET
604
      FORMAT(A1)
C
```

```
IF (ANSETLEQ.'Y') THEN
       WRITE(*,239)UL
239
       FORMAT( : FNTFR FPOXY THICKNESS ('.A2.'): ".2X.')
       READ *.DELE
     ELSEIF (ANSETLEQ.'N') THEN
       GOTO 25
     ELSE
       GOTO 26
25
      ENDIF
\mathbf{C}
28
    CALL CLS
    WRITE(*,605)
    FORMAT(alle)
605
    PRINT *, THE CURRENT ENTRY FOR THERMAL CONDUCTIVITY IS ',KE
  +. T.UK
    WRITE(*.240)
   FORMAT(/,"
              WOULD YOU LIKE TO CHANGE THE THERMAL CONDUCTIVIT
  +Y? (Y OR N) 1.2X.)
    READ(*,606)ANSFK
606
    FORMAT(A1)
(
     IF (ANSEKJEQTY) THEN
       WRITE(*.241)UK
241
       FORMAT( .: ENTER EPOXY THERMAL CONDUCTIVITY (",A12.")
  +: '.2X.)
       READ *.KE
     ELSEIF (ANSEK.EQ.'N') THEN
       GOTO 27
     ELSE
       GOTO 28
     ENDIF
\mathbf{C}
C PROVIDE ANOTHER REVIEW OF EPOXY ENTRIES
\mathbf{C}
27 CALL CLS
  WRITE(*,242)
242 FORMAT(#//.
              YOU HAVE MADE THE FOLLOWING CORRECTIONS TO THE EP
  +OXY ENTRIES.™)
  GOTO 32
(`
  ELSEIF (ANSE.EQ.N.) THEN
   GOTO 29
  ELSE
   GOTO 30
  ENDIF
(
(
(
(
29 CALL CLS
   WRITE(*,243)
```

```
+***-****'. )
(
944 WRITE(*,244)UL
244 FORMAT(* ENTER CARRIER LENGTH (",A2,"): 1.2X,()
   RFAD *.CL
(:
(
  CARRIER LENGTH MUST BE GREATER THAN OR EQUAL TO SUBSTRATE LENGTH
   IF (CLLT.SL) THEN
     CALL CLS
     WRITE(*,243)
     PRINT *, CARRIER LENGTH MUST BE GREATER THAN OR EQUAL TO SUBS
   +TRATE LENGTH.
     GOTO 944
   ENDIF
\mathbf{C}
C
   WRITE(*,246)UL
246 FORMAT(/,' ENTER CARRIER THICKNESS (',A2,'): ',2X,\)
   READ *.DELC
C
   WRITE(*,247)UK
247 FORMAT(\(\times\) ENTER CARRIER THERMAL CONDUCTIVITY (\(\times\). \(\times\).
   READ *.KC
(,
38 CALL CLS
   CW = SW
€,
C REVIEW CARRIER ENTRIES
C
   WRITE(*.248)
248 FORMAT(##,*
                YOU HAVE MADE THE FOLLOWING ENTRIES FOR THE CARRI
   +ER.;/)
                            `.CL
36 PRINT ( L) LENGTH
   PRINT *."
             2.) WIDTH
                            ,CW
   PRINT *."
             3.) THICKNESS
                            ,DELC
   PRINT *."
                         '.KC
            4.) k
   PRINT *
   PRINT *: CHANGING WIDTH WILL ALSO CHANGE SUBSTRATE AND EPOXY W
   +IDTHS.
C
   WRITE(*,249)
249 FORMAT(/,'
                DO YOU WISH TO MAKE ANY CHANGES? (Y OR N) ',2X,\)
   READ(*,607)ANSC
607 FORMAT(A1)
(,
 MAKE CORRECTIONS OR CHANGES TO CARRIER ENTRIES
C
(
   TE (ANSC.EQ. Y') THEN
33
     CALL CLS
     WRITE(*,608)
608
      FORMAT(\#)
933
      PRINT 1. THE CURRENT ENTRY FOR LENGTH IS 1.CL, 1.UL
```

```
WRITE(*,250)
250
      FORMAT(/,
                   WOULD YOU LIKE TO CHANGE LENGTH? (Y OR N) ',2X,
   +1)
     READ(*,609)ANSCL
609
      FORMAT(A1)
(
       IF (ANSCLEQ.Y') THEN
         WRITE(*.251)UL
251
         FORMAT( ... ENTER CARRIER LENGTH (',A2.'): ',2X.\(\)
         READ *.CL
         HF (CLLLISL) THEN
          CALL CLS
          WRITE(*,610)
610
           FORMAT(iiii)
          PRINT *: CARRIER LENGTH MUST BE GREATER THAN OR EQ
   +UAL TO SUBSTRATE LENGTH."
          PRINT *. THE PRESENT ENTRY FOR SUBSTRATE LENGTH IS
   + ".SL." ..UT
          PRINT *
          GOTO 933
         ENDIF
C
       ELSEIF (ANSCL.EQ.'N') THEN
        IF (CL.I T.SL) THEN
          CALL CLS
          WRITE(*,611)
611
           FORMAT(///)
          PRINT *, CARRIER LENGTH MUST BE GREATER THAN OR EQ
   +UAL TO SUBSTRATE LENGTH.
          PRINT *.' THE PRESENT ENTRY FOR SUBSTRATE LENGTH IS
   + ".SL.UL
          PRINT *
          GOTO 933
        ENDIF
        GOTO 39
       FLSE
        GOTO 33
       ENDIF
(
39
     CALL CLS
     WRITE(*,612)
612
     FORMAT(#/)
     PRINT *: THE CURRENT ENTRY FOR WIDTH IS ",CW." ,UL
     PRINT *: CHANGING THIS ENTRY WILL CHANGE SUBSTRATE AND
     PRINT *, EPOXY WIDTHS. THEY ARE ALL EQUAL.
     WKITE(*,252)
252
     FORMAT(/,
                  WOULD YOU LIKE TO CHANGE WIDTH? (Y OR N) 7,2X,\
   +)
     READ(*,613)ANSCW
613
     FORMAT(A1)
C
       IF (ANSCW.EQ.'Y') THEN
        WRITE(*,253)UL
253
         FORMAT(/,' ENTER CARRIER WIDTH ('.A2.'): ',2X,-)
        READ *.CW
```

```
SW = CW
        EW = CW
       ELSEIF (ANSCW.EQ.'N') THEN
        GOTO 34
       ELSE
        GOTO 39
       ENDIF
C
34
     CALL CLS
     WRITE(*,614)
614
      FORMAT(##/)
     PRINT *, THE CURRENT ENTRY FOR THICKNESS ',DELC,' ',UL
     WRITE(*,254)
254
     FORMAT(::
                  WOULD YOU LIKE TO CHANGE THICKNESS? (Y OR N) ;
   +2X_{\odot}
     READ(*,615)ANSCT
615
      FORMAT(A1)
C
       IF (ANSCT.EQ.'Y') THEN
        WRITE(*.2551)UL
2551
         FORMAT( ... ENTER CARRIER THICKNESS ("A2,"): ",2X,:)
        READ *.DELC
       ELSEIF (ANSCT.EQ.N.) THEN
        GOTO 35
       ELSE
        GOTO 34
       ENDIF
\mathbf{C}
35
     CALL CLS
     WRITE(*,616)
616
      FORMAT(////)
     PRINT *.' THE CURRENT ENTRY FOR THERMAL CONDUCTIVITY IS ',KC
   +,``,UK
     WRITE(*,256)
                  WOULD YOU LIKE TO CHANGE THE THERMAL CONDUCTIVIT
     FORMAT(/,`
   +Y? (Y OR N) (2X,\)
     READ(*.617)ANSCK
617
      FORMAT(A1)
C
       IF (ANSCK.EQ.'Y') THEN
         WRITE(*,257)UK
         FORMAT(); ENTER CARRIER THERMAL CONDUCTIVITY (".A12,
   +1): 1.2X...)
         READ *.KC
       ELSEIF (ANSCK.EQ.'N') THEN
        GOTO 40
       ELSE
        GOTO 35
       ENDIF
(
  ALLOW FOR ANOTHER REVIEW OF CARRIER ENTRIES
(
40
   CALL CLS
   WRITE(*,258)
                  YOU HAVE MADE THE FOLLOWING CORRECTIONS TO THE C
258 FORMAT(iiii)
```

```
+ARRIER ENTRIES.://)
  GOTO 36
C
(
  ELSEIF (ANSC.EQ.N') THEN
   GOTO 37
  ELSE
   GOTO 38
37
  ENDIF
\mathbf{C}
C
C
\mathbf{C}
  CALL CLS
  WRITE(*,260)
+*********
  +********(//)
C
(
 INITIAL TEMPERATURE
(
  WRITE(*,261)UT
261 FORMAT(" ENTER THE INITIAL CHIP TEMPERATURE ("A1,"). "(2X,\)
  READ *.IT
C UPPER AMBIENT TEMPERATURE
  WRITE(*,262)UT
262 FORMAT(). ENTER THE UPPER SURFACE AMBIENT TEMPERATURE ('A1,'
  +). (2X.)
  READ *,UPRT
C
C LOWER AMBIENT TEMPERATURE
C
  WRITE(*.263)UT
263 FORMAT(/, ENTER THE LOWER SURFACE AMBIENT TEMPERATURE (',A1.'
  +). (2X, 1)
  READ *,LWRT
\mathbf{C}
C RIGHT SIDE AMBIENT TEMPERATURE
(
  WRITE(*,264)UT
264 FORMAT()." ENTER THE RIGHT SURFACE AMBIENT TEMPERATURE ('.A.L.'
  +). (.2X.)
  READ *.RT
 LEFT SIDE AMBIENT TEMPERATURE
(
  WRITE(*,265)UT
```

```
265 FORMATO:
                 ENTER THE LEFT SURFACE AMBIENT TEMPERATURE (',A1,')
   +. `,2X,\)
   READ *,LT
(
  FRONT AMBIENT TEMPERATURE
(
   WRITE(*,266)UT
266 FORMAT(-
                ENTER THE FRONT SURFACE AMBIENT TEMPFRATURE (',A1,'
   +). '.2X.)
   READ *.FT
C
  REAR SIDE AMBIENT TEMPERATURE
(
\mathbf{C}
   WRITE(*,267)UT
267 FORMAT(4)
                ENTER THE REAR SURFACE AMBIENT TEMPERATURE (',A1,')
   +...,2X.
   READ * BT
(
C
\mathbf{C}
  REVIEW TEMPERATURE ENTRIES
C
50 CALL CLS
   WRITE(*,268)
268 FORMAT(////,
                  YOU HAVE MADE THE FOLLOWING AMBIENT TEMPERATURE
   +ENTRIES.',)
   PRINT *,*
             1.) INITIAL CHIP TEMPERATURE
                                               ',IT,' ',UT
   PRINT *.'
             2.) UPPER AMBIENT TEMPERATURE
                                                  ',UPRT,' ',UT
   PRINT *."
                                                   ',LWRT,' ',UT
             3.) LOWER AMBIENT TEMPERATURE
   PRINT *C
             4.) RIGHT AMBIENT TEMPERATURE
                                                  ',RT,' ',UT
   PRINT *.
             5.) LEFT AMBIENT TEMPERATURE
                                                 ',LT,' ',UT
   PRINT *."
             6.) FRONT AMBIENT TEMPERATURE
                                                  ',FT,' ',UT
   PRINT *C
             7.) REAR AMBIENT TEMPERATURE
                                                  `,BT,` ',UT
(
(
   WRITE(*,269)
269 FORMAT(/,
                 WOULD YOU LIKE TO MAKE ANY CORRECTIONS? (Y OR N)
   +.2X.
   READ(*.618)ANST
618 FORMAT(A1)
C
(
  MAKE CORRECTIONS OR CHANGES TO TEMPERATURE ENTRIES
\mathbf{C}
   IF (ANST.EQ.'Y') THEN
42
     CALL CLS
     WRITE(*,268)
     PRINT *.:
                                                 '.IT.' '.UT
               1.) INITIAL CHIP TEMPERATURE
     PRINT *."
                                                    ',UPRT,' ',UT
               2.) UPPER AMBIENT TEMPERATURE
     PRINT *."
                                                    ',LWRT,' ',UT
               3.) LOWER AMBIENT TEMPERATURE
     PRINT *."
                                                   '.RT.' ',UT
               4.) RIGHT AMBIENT TEMPERATURE
     PRINT *.:
                                                   ',LT,' ',UT
               5.) LEFT AMBIENT TEMPERATURE
     PRINT *."
               6.) FRONT AMBIENT TEMPERATURE
                                                    ',FT,' ',UT
                                                   ',BT,' '.UT
     PRINT *.
               7.) REAR AMBIENT TEMPERATURE
(
     WRITE(*.2682)
                    WHICH TEMPERATURE WOULD YOU LIKE TO CHANGE?"./
2682
     FORMAT(#/.
```

```
+. SELECT ZERO THROUGH SEVEN. [2X.)
     READ(*.618)DELT
C CORRECT INITIAL TEMPERATURE
     IF (DELT.EQ.T) THEN
       CALL CLS
       WRITF(*,270)
270
       FORMAT(%)
       PRINT *. INITIAL CHIP TEMPREATURE IS ',IT,' ',UT
       WRITE(*,271)UT
271
       FORMAT(), ENTER THE NEW VALUE.(',A1,') ',2X,\)
       READ *JT
C
 CORRECT UPPER AMBIENT TEMPERATURE
     ELSEIF (DELT.EQ.2') THEN
       CALL CLS
       WRITE(*,620)
620
       FORMAT(...)
       PRINT *." UPPER AMBIENT TEMPERATURE IS LUPRT, LUT
       WRITE(*,273)UT
273
       FORMAT( .
                    ENTER THE NEW VALUE.(',A1.') \( \).2X,\(\)
       READ *, UPRT
(
C CORRECT LOWER AMBIENT TEMPERATURE
     ELSEIF (DELT.EQ.3') THEN
       CALL CLS
       WRITE(*,623)
623
       FORMAT(%)
       PRINT *, LOWER AMBIENT TEMPERATURE IS 'LWRT,' ',UT
       WRITE(*,275)UT
275
       FORMAT(/,' ENTER THE NEW VALUE.(',A1,') ',2X,\)
       READ *.LWRT
C
C CORRECT RIGHT AMBIENT TEMPERATURE
     ELSEIF (DELT.EQ.'4') THEN
       CALL CLS
       WRITE(*.626)
       FORMAT(%)
626
       PRINT *: RIGHT AMBIENT TEMPERATURE IS '.RT.' '.UT
       WRITE(*.277)UT
277
       FORMAT(\cdot, \cdot)
                    ENTER THE NEW VALUE (LALT) (L2X.)
       READ * RT
  CORRECT LEFT AMBIENT TEMPERATURE
     ELSEIF (DELT.EQ '5') THEN
       CALL CLS
       WRITE(*,629)
624
       FORMAT(#)
       PRINT ** LEFT AMBIENT TEMPFRATURE IS LEET LUT
       WRITE(*.279)UT
279
       FORMAT( : ENTER THE NEW VALUE (".AL") (".2X.")
       READ *.LT
C CORRECT FRONT AMBIENT TEMPFRATURE
     ELSEIF (DELT.EQ.'6') THEN
```

```
CALL CLS
       WRITE(*,632)
632
       FORMAT(a)
       PRINT ** FRONT AMBIENT TEMPERATURE IS SET, SUT
       WRITE(*,281)UT
       FORMAT(C) ENTER THE NEW VALUE (LALT) (L2X.)
281
       READ *JT
(
C CORRECT REAR AMBIENT TEMPERATURE
     ELSEIF (DELT.EQ.7') THEN
       CALL CLS
       WRITE(*,635)
635
       FORMAT(///)
       PRINT *," REAR AMBIENT TEMPERATURE IS ',BT,' ',UT
       WRITE(*,284)UT
       FORMAT(;' ENTER THE NEW VALUE.(',A1,') ',2X.\)
284
       READ *,BT
C
  MAKE NO CHANGES
     ELSEIF (DELT.EQ.'0') THEN
          CALL CLS
4201
       WRITE(*,6320)
        FORMAT(///, YOU HAVE DECIDED TO MAKE NO CORRECTIONS! /,
6320
       IS THIS CORRECT? (Y OR N) \(\frac{1}{2}X\tau\)
       READ(*,618) ANSTI
       IF (ANSTLEQ.'Y') THEN
         GOTO 51
       ELSEIF (ANSTLEQ.N') THEN
         GOTO 42
       ELSE
         GOTO 4201
       ENDIF
C
     ELSE
       GOTO 42
     ENDIF
C
C ALLOW FOR ANOTHER REVIEW OF TEMPERATURE ENTRIES
\mathbf{C}
48
     CALL CLS
     WRITE(*,635)
     WRITE(*,285)
      FORMAT(##, WOULD YOU LIKE TO MAKE ANY MORE CORRECTIONS)...
285
   + OR REVIEW TEMPERATURE ENTRIES. (Y OR N) 1.2X.)
     READ(*,618)ANSTL
     IF (ANSTL.EQ.'Y') THEN
       GOTO 42
     ELSEIF (ANSTLEQ.N') THEN
       GOTO 51
     ELSE
       GOTO 48
     ENDIF
(
    ELSFIF (ANST.EQ.N.) THEN
     GOTO 51
```

```
ELSE
    GOTO 50
   ENDIF
(
(
(
       (
(
 PROVIDE A CORRELATION BETWEEN NODE NUMBERRS AND MATRIX LOCATION
(
51 \quad NUM = 1
   DO 60.1 = 1.NPL/NWIDE
    DO 61 J = 1, NWIDE:
     JH(NUM) = J
     NUM = NUM + 1
    CONTINUE
61
60 CONTINUE
C
  NUMA = 1
  CH = 0
  H = 1
   DO 62.1 = 1.NPL
    IH(NUMA) = H
    CH = CH + 1
    IF (CH.EQ.NWIDE) THEN
     H = H + 1
     CH = 0
    ENDIF
   NUMA = NUMA + 1
62 CONTINUE
 PROVIDE ALTERNATIVE HEAT INPUT METHODS
(
(
  CALL CLS
   WRITE(*,290)
290 FORMAT("*******
  HEAT INPUT TO THE MICROCIRCUIT OCCURS ONLY ON THE UPPER 1/4.
      SUBSTRATE SUFACE. HEAT INPUT IS ACCOMPLISHED BY ONE OF 1/4.
      THE FOLLOWING METHODS: #.
      1.) ENTER AS A TOTAL HEAT APPLIED TO THE CHIP. 7.
      2.) ENTER AS AVERAGE HEAT PER UNIT AREA. 7.
      3.) ENTER HEAT NODE BY NODE. 7.
      4.) NO HEAT INPUT. J/,
      PLEASE SELECT A NUMBER ONE THROUGH FOUR. (2X.)
  READ(*.304)SELH
(
  HF (SELFLEQ.T.OR.SELH.EQ.2.OR.SELH.EQ.3.OR.SELH.EQ.4.) THEN
64
    WRITE(*,291)SELH
```

```
FORMAT(#; YOU HAVE SELECTED NUMBER ".A1." OF FOUR ALTERNAT
   +IVES. ',,' IS THIS THE CORRECT CHOICE? (Y OR N) '\2X\\)
     READ(*,304)ANSH
\mathbf{C}
     IF (ANSH.EQ.'Y') THEN
      GOTO 63
     ELSEIF (ANSH.EQ.'N') THEN
      GOTO 51
     ELSE
       CALL CLS
       WRITE(*,290)
       GOTO 64
     ENDIF
63
   ELSE
     GOTO 51
   ENDIF
C
  DETERMINE UNITS FOR HEAT INPUT
   IF ((ANSN.EQ.'E').AND.(SELILEQ.'2')) THEN
     UH = Btu(hr*in^2)
     UAH = 'Btu/hr'
   ELSEIF ((ANSN.EQ.'E').AND.(SELH.EQ. T.OR.SELH.EQ.'3')) THEN
     UH = '(Btu) / (hr)
   ELSEIF ((ANSN.EQ.'S').AND.(SELH.EQ.'2')) THEN
     UH = WATTS/(cm^2)
     UAH = 'WATTS'
   ELSEIF ((ANSN.EQ.'S').AND.(SELH.EQ.'1'.OR.SELH.EQ.'3')) THEN
     UH = WATTS
   ENDIF
C
\mathbf{C}
C ALLOW FOR RE-SELECTION OF HEAT INPUT METHOD OR CONTINUE WITH
C
  INITIAL SELECTION
C
  CHOICE #1
C
(
   CALL CLS
   IF (SELH.EQ.'T) THEN
65
      WRITE(*,292)UH
                  YOU HAVE SELECTED TO INPUT HEAT AS A TOTAL HEAT
292
      FORMAT(/%,*
       APPLIED TO THE SURFACE. V.
   + ENTER TOTAL HEAT APPLIED TO THE SURFACE (LA13.) L2X.()
     READ *,THEAT
66
      WRITE(*,293)
293
      FORMAT(A^* IS THIS THE CORRECT ENTRY (Y OR N) (2XA)
     READ(*,304)ATH
C MAKE HEAT ENTRY AND ALLOW FOR CORRECTION
     IF (ATH.EQ.'Y') THEN
       THPN = THEAT/NPL
       PRINT *
       PRINT *. TOTAL HEAT PER NODE IS "THPN." "UH
```

```
C
  FILL HEAT MATRIX WITH DESIRED VALUES
(
       DO 80 I = I.NDEEP
        DO 81 J = 1.NWIDE
          HEAT(LJ) = THPN
         CONTINUE
81
       CONTINUE:
80
\mathbf{C}
     ELSEIF (ATH.EQ.'N') THEN
       CALL CLS
       GOTO 65
     ELSE
       CALL CLS
       WRITE(*,294)
294
       FORMAT(////)
       PRINT *,' TOTAL HEAT APPLIED TO THE SURFACE IS ',THEAT,'
   + ',UH
       GOTO 66
     ENDIF
\mathbf{C}
(
  CHOICE #2
C
   ELSEIF (SELH.EQ.27) THEN
67
     WRITE(*,295)UH
      FORMAT(m/C YOU HAVE SPLECTED TO ENTER THE AVERAGE HEAT OV
   +ER THE'J.
   + UPPER SUBSTRATE SURFACE. #.
   +' ENTER THE DESIRED HEAT INPUT (',A13,'), ',2X,\)
     READ *,AHI:AT
C. MAKE ENTRY AND ALLOW FOR CORRECTION
\mathbf{C}
     WRITE(*,296)
68
296
      FORMAT(//, IS THIS THE CORRECT ENTRY. (Y OR N) ',2X,\)
     READ(*,304)ANSHA
C
     IF (ANSHA.EQ.'Y') THEN
       THPN = AHEAT*SL*SW/NPL
       PRINT *
       PRINT *, TOTAL HEAT PER NODE IS ',THPN,' ',UAH
C
(
  FILL HEAT MATRIX WITH DESIRED VALUES
(
     DO 82 I = 1, NDEEP
       DO 83 J = I.NWIDF
        HEAT(I,J) = THPN
83
       CONTINUE
82
     CONTINUE
     ELSEIF (ANSHALEQIN') THEN
       CALL CLS
       GOTO 67
     ELSE
       CALL CES
```

```
WRITE(*,294)
      PRINT *.* AVERAGE HEAT OVER SUBSTRATE SURFACE IS ",AHEAT.
   + .UH
      GOTO 68
     ENDIF
(
C
  CHOICE #3
(
   ELSEIF (SELILEQ.3') THEN
70
     WRITE(*,297)
     FORMAT(
                 YOU HAVE SELFCIED TO ENTER HEAT NO DALLY M.
     ENTER THE TOTAL NUMBER OF NODES DESIGNATED FOR HEAT INPUT.
   +.2X_{\odot}
     READ *,TOTNOD
C
C. THIS IS NODE BY NODE, GET NUMBER OF ENTRIES THEN LOOP UNTIL ALL
C ENTRIES HAVE BEEN MADE
C
  TELL USER MAXIMUM ENTRIES POSSIBLE
     IF (TOTNOD.GT.NPL) THEN
      PRINT *
      PRINT *; MAXIMUM ENTRY IS ',NPL
      CALL CLS
      GOTO 70
     ENDIF
(
(`
  MAKE ENTRIES
(
     DO 71.1 = 1.TOTNOD
      NC = 1
       CALL CLS
       WRITE(*,402)NC.TO1NOD
402
       FORMAT( ... THIS IS NUMBER 113.1 OF 1.13.1 ENTRIES')
       WRITE(*,298)
298
       FORMAT(: ENTER THE NODE NUMBER FOR HEAT INPUT. 1,2X.))
       READ *,NN
      IF (NN.EQ.0.OR.NN.GT.NPL) THEN
        GOTO 75
       ENDIF
       WRITE(*,299)UH
299
       FORMAT(/,' ENTER HEAT INPUT (',A13,'). ',2X.\(\))
       READ *,NHEAT
       HEAT(IH(NN),JH(NN)) = NHEAT
71
     CONTINUE
(
C PROVIDE OPPORTUNITY FOR CORRECTIONS OR FURTHER ENTRIES
\mathbf{C}
73
     CALL CLS
     WRITE(*,401)TOTNOD
401
      FORMAT( %: YOU HAVE MADE '.13.' NODAL ENTRIES.')
     WRITE(*,400)
     FORMAT(: DO YOU WISH TO MAKE ANY MORE ENTRIES OR CORRECTIO
400
   +NS2 (Y OR N) (2X.)
     READ(*.304)AHN
```

```
\mathbf{C}
    IF (AHN.EQ.'Y') THEN
     CALL CLS
     GOTO 70
    ELSEIF (AHN.EQ.'N') THEN
     GOTO 72
    ELSE
     GOTO 73
72
    ENDIF
(
\mathbf{C}
C NO HEAT INPUT - CHOICE #4 - HEAT MATRIX STAYS INITIALIZED AT ZERO
  ELSEIF (SELH.EQ. 4') THEN
   GOTO 86
  ENDIF
(
86
  CONTINUE
0
(
        ('***
        C
(
 DETERMINE INCREMENTAL MEASUREMENTS IN THE X AND Y DIRECTIONS
(
  DELX = SL/NDEEP
  DELY = SW/NWIDE
(
C LEFT OR RIGHT EDGE TO OUTSIDE
  SYLR = 2 * KS * DELX * DELS / DELY
C FRONT OR BACK TO OUTSIDE
  SXFB = 2 * KS * DFLY * DELS DELX
CINNER MATRIX MOVEMENT IN THE Y DIRECTION
  SYY = KS * DELX * DELS / DELY
C INNER MATRIX MOVEMENT IN THE X DIRECTION
  SXX = KS * DELY * DELS / DELX
C SUBSTRATE TO TOP SURFACE
  SZT = 2 * KS * DELX *DELY : DELS
(`
C SUBSTRATE TO EPOXY
  SZE = 2 * DELX * DELY / ((DELS/KS) + (DELE/KE))
(
C. LEFT OR RIGHT EDGE TO OUTSIDE
  EYLR = 2*KE*DELX*DELE/DELY
C FRONT OR BACK EDGE TO OUTSIDE
```

```
EXFB = 2*KE*DFLY*DELE/DELX
(
(
  INNER MATRIX MOVEMENT IN THE Y DIRECTION
   EYY = KE*DELX*DELE DELY
(
C - FPONY TO CARRIER
   EZC = 2*DELX*DELY*((DELE/KE)+(DELC KC))
(
C. INNER MATRIX MOVEMENT IN THE X DIRECTION
   EXX = KE*DELE*DELY/DELX
C****************GENERATE CONSTANTS FOR INTERIOR CARRIER LAYER*******
(
C LEFT OR RIGHT EDGE TO OUTSIDE
   CYLR = 2*KC*DELX*DELC/DELY
C
C INNER MATRIX MOVEMENT IN THE Y DIRECTION
   CYY = KC*DELX*DELC/DELY
(
C. INNER MATRIX MOVEMENT IN THE X DIRECTION
   CXX = KC*DELC*DELY:DELX
(
  CARRIER TO BOTTOM OUTER EDGE
( .
   CZB = 2*KC*DELX*DELY DELC
( '
(
 CARRIER TO FRONT EDGE IF NO EAR EXISTS
   CXFB = 2*KC*DELY*DELC/DELX
(
\mathbf{C}
C****COEFFICIENTS FOR SUBSTRATE, EPOXY, AND INTERIOR CARRIER LAYERS*****
( `*******************************
C
C
   DO 90 I = LNPL
   N = 1
   IB = NPL + I
   ID = 2*NPL+I
\mathbf{C}
     C*
(
(
    IF((IH(I).EQ.1.OR.IH(I).EQ.NDEEP).AND.(JH(I).EQ.1.OR.JH(I).EQ.
   +NWIDE)) THEN
(
  DETERMINE CONNECTIONS FOR TOP LAYER
      IF (HEAT(IH(I).JH(I)).EQ.0.0) THEN
        NCON(LN) = 6
      ELSE
        NCON(I.N) = 1
      ENDIF
  CONNECTIONS FOR EPOXY LAYER
      NCON(IB,N) = 6
```

á

```
C
(
  LEFT AND RIGHT COEFFICIENTS DEPENDING ON WHICH EDGE
(
(
     TEFT EDGE
       HF (JH(I),I-Q,I) THI-N
(
      TITT COFFFICIENT
         COFF(LN) = SYLR
         COEF(BN) = EYLR
         N = N + 1
         NCON(LN) = 7551
         NCON(IB.N) = 7551
      RIGHT COEFFICIENT
(
         COEF(LN) = SYY
         COFF(IB.N) = FYY
         N = N + 1
         NCON(I,N) = 10*(I+1)+1
         NCON(IB,N) = 10*(IB+1)+1
(
      RIGHT EDGE
       ELSEIF (JH(I).EQ.NWIDE) THEN
      LEFT COEFFICIENT
(
         COEF(LN) = SYY
         COEF(IB,N) = EYY
         N = N + 1
         NCON(I,N) = 10*(I-1)+1
         NCON(IB.N) = 10*(IB-1)+1
(
      RIGHT COFFFICIENT
         COEF(I,N) = SYLR
         COEF(IB.N) = EYLR
         N = N + 1
         N(ON(LN) = 7541
         NCON(IB.N) = 7541
        ENDIF
(
   FROM LAND BACK COFFFICIENTS DEPENDING ON WHICH EDGE
(
      FRONT EDGE
(
        IF (IH(I).EQ.1) THEN
       FRONT COEFFICIENT
         COFF(LN) = SXFB
         COEF(IB.N) = EXFB
          N = N + 1
          NCON(LN) = 7521
          NCON(IB.N) = 7521
       BACK COEFFICIENT
 (
          COEF(I.N) = SXX
          COEF(IB,N) = EXX
          N = N + 1
          NCON(I,N) = 10*(I+NWIDE)+1
          NCON(IB,N) = 10*(IB+NWIDE)+1
 (
       BACK EDGE
        FLSFIF (IH(I).EQ.NDEEP) THEN
 (
       FRONT COEFFICIENT
          COEF(I,N) = SXX
          COEF(IB.N) = EXX
          N = N + 1
          NCON(I,N) = 10*(I-NWIDE)+1
```

```
NCON(IB,N) = 10*(IB-NWIDE)+1
C
     BACK COEFFICIENT
        COEF(I,N) = SXFB
        COEF(IB,N) = EXFB
        N = N + 1
        NCON(I,N) = 7531
        NCON(IB.N) = 7531
      ENDIF
  TOP COEFFICIENT
      COEF(I,N) = SZT
      COFF(IP.N) = SZE
      N = N + 1
      NCON(LN) = 7511
      NCON(IB.N) = 10*I+1
C BOTTOM COEFFICIENT
      COFF(I,N) = SZE
      COEF(IB,N) = EZC
      N = N + 1
      NCON(I,N) = 10^*(I+NPL)+1
      NCON(IB.N) = 10*(IB+NPL)+1
C. HEAT INPUT
      HF (HEAT(IH(I),JH(I)).NE.0.0) THEN
        COEF(I,N) = HEAT(IH(I),JH(I))
        N = N + 1
        NCON(I.N) = 9991
      ENDIF
(
C
()
(
     ELSFIF ((III(I).EQ.1.OR.IH(I).EQ.NPLEP).AND.(JH(I).NE.1.OR.JH(I)
   +.NE.NWIDL ;; THEN
(
C. DETERMINE NUMBER OF CONNECTIONS FOR SUBSTRATE LAYER
      IF (HEAT(IH(1),JH(1)),EQ.0.0) THEN
        NCON(I.N) = 6
      1:1.SE
        NCON(LN) = 7
      FNDIF
C. DETERMINE NUMBER OF CONNECTIONS FOR EPOXY LAYER
      NCON(IB,N) = 6
C. LEFT COEFFICIENT
      COEF(I,N) = SYY
      COEF(IB,N) = EYY
      N = N + 1
       NCON(I,N) = 10*(I-1)+1
       NCON(IB,N) = 10*(IB-1)+I
C RIGHT COEFFICIENT
      COEF(I,N) = SYY
      COEF(IB,N) = EYY
       N = N + 1
```

```
NCON(1,N) = 10^{8}(1+1)+1
       NCON(IB,N) = 10^{\circ}(IB+i)+1
  TRONT AND BACK COFFFICIENTS DEPENDENT ON WHICH EDGE
(
      FRONT EDGE
       IF (JH(I).EQ.1) THEN
(
      FRONT COEFFICIENT
         COEF(LN) = SXFB
         COEF(IB.N) = EXFB
         N = N + 1
         NCON(I,N) = 7521
         NCON(IB,N) = 7521
C
      BACK COEFFICIENT
         COEF(I,N) = SXX
         COEF(IB.N) = EXX
         N = N + 1
         NCON(I.N) = 10*(I+NWIDE)+1
         NCON(IB,N) = 10*(IB+NWIDE)+1
(`
      BACK EDGE
       ELSFIF (III(I).FQ.NDEEP) THEN
(
      FRONT COEFFICIENT
         COId(LN) = SXX
         COEF(IB.N) = EXX
         N = N + 1
         NCON(LN) = 10^{\circ}(I-NWIDE) + E
         NCON(IB.N) = 10^{\circ}(IB-NWID1) + 1
(
      BACK COLIFICIENT
         COFF(I,N) = SXFB
         COEF(IB,N) = EXFB
         N = N + 1
         NCON(I,N) = 7531
         NCON(IB,N) = 7531
       ENDIF
C TOP COEFFICIENT
       COFF(LN) = SZT
       COEF(IB.N) = SZE
       N = N + 1
       NCON(I,N) = 7511
       NCON(IB,N) = 10*I+1
  BOTTOM COEFFICIENT
( `
       COEF(LN) = SZE
       COEF(IB.N) = EZC
        N = N + 1
        NCON(I,N) = 10*(I+NPL)+I
        NCON(IB,N) = 10*(IB+NPL)+1
C. THEAT INPUT
       IF (HEAT(IH(I).JH(I)).NE.0.0) THEN
         COEF(LN) = HEAT(IR(I)JH(I))
         N = N + 1
         NCON(I,N) = 9991
        ENDIF
C
```

```
(
( **
   *********LEFT AND RIGHT EDGES EXCLUDING CORNERS**************
(
(
     ELSEIF ((JH(I).FQ.1.OR.JH(I).EQ.NWIDE).AND.(IH(I).NE.1.OR.IH(I)
   +.NE.NDEEP)) THEN
(
C. DETERMINE NUMBER OF CONNECTIONS FOR SUBSTRATE LAYER
       IF (HEAT(H(1),JH(1)),NE.0.0) THEN
         NCON(1,N) = 7
       ELSE
         NCON(I.N) = 6
       ENDIF
(
C. DETERMINE THE NUMBER OF CONNECTIONS FOR EPOXY AND CARRIER LAYERS
       NCON(IB.N) = 6
       NCON(ID,N) = 6
C
C
\mathbf{C}
  LEFT AND RIGHT COEFFICIENTS DEPENDING ON WHICH EDGE
C
      LEFT EDGE
       IF (JH(I).EQ.1) THEN
C
      LEFT COEFFICIENT
         COEF(I.N) = SYLR
         COEF(IB.N) = EYLR
        COEF(ID,N) = CYLR
         N = N + 1
         NCON(I.N) = 7551
         NCON(IB,N) = 7551
         NCON(ID,N) = 7551
(
      RIGHT COEFFICIENT
        COEF(I,N) = SYY
         COEF(IB.N) = EYY
         COEF(ID.N) = CYY
         N = N + 1
         NCON(I,N) = 10*(I+1)+1
         NCON(IB,N) = 10*(IB+1)+1
         NCON(ID.N) = 10*(ID+1)+1
C
      RIGHT EDGE
      ELSEIF (JH(I).EQ.NWIDE) THEN
(
     LEFT COEFFICIENT
        COEF(I,N) = SYY
         COEF(IB,N) = EYY
        COEF(ID.N) = CYY
         N = N + 1
         NCON(I,N) = 10*(I-1)+1
         NCON(IB,N) = 10*(IB-1)+1
         NCON(ID.N) = 10*(ID-1)+1
(
     RIGHT COEFFICIENT
        COEF(I,N) = SYLR
        COEF(IB,N) = EYLR
        COEF(ID.N) = CYLR
        N = N + 1
         NCON(I.N) = 7541
        NCON(IB,N) = 7541
```

```
NCON(ID,N) = 7541
       ENDIF
C
C
    FRONT COEFFICIENT
       COEF(I.N) = SXX
       COEF(IB,N) = EXX
       COEF(ID,N) = CXX
       N = N + 1
       NCON(I,N) = 10*(I-NWIDE)+1
       NCON(IB,N) = 10*(IB-NWIDE)+1
       NCON(ID.N) = 10*(ID-NWIDF)+1
(
    BACK COEFFICIENT
       COEF(I.N) = SXX
       COEF(IB.N) = FXX
       COEF(ID.N) = CXX
       N = N + 1
       NCON(I.N) = 10*(I+NWIDE)+1
       NCON(IB,N) = 10*(IB+NWIDE)+1
       NCON(ID,N) = 10*(ID+NWIDE)+1
C
C
    TOP COEFFICIENT
       COEF(I,N) = SZT
       COEF(IB,N) = SZE
       COEF(ID.N) = EZC
       N = N + 1
       NCON(I,N) = 7511
       NCON(IB,N) = 10*I+1
       NCON(ID,N) = 10*IB+1
C
C
    BOTTOM COEFFICIENT
       COEF(I,N) = SZE
       COEF(IB,N) = EZC
       COEF(ID,N) = CZB
       N = N + 1
       NCON(LN) = 10*(I+NPL)+1
       NCON(IB,N) = 10*(IB+NPL)+1
       NCON(ID.N) = 7561
(
    HEAT INPUT
       IF (HEAT(IH(I),JH(I)),NE.0.0) THEN
         COEF(I.N) = HFAT(IH(I)JH(I))
         N = N + 1
         NCON(I,N) = 9991
       ENDIF
C
C
C*****DETERMINE COEFFICIENTS FOR ALL NODES NOT TOUCHING AN EDGE*******
\mathbf{C}
\mathbf{C}
     ELSEIF ((IH(I).NE.1.OR.IH(I).NE.NDEEP).AND.(JH(I).NE.1.OR.JH(I)
   +.NE.NWIDE)) THEN
(
 DETERMINE NUMBER OF CONNECTIONS FOR SUBSTRATE LAYER
       IF (HEAT(IH(I),JH(I)).NE.0.0) THEN
```

```
NCON(1,N) = 7
       ELSE
        NCON(I,N) = 6
       ENDIF
(
  DETERMINE NUMBER OF CONNECTIONS FOR EPOXY AND CARRIER LAYERS
(`
       NCON(IB,N) = 6
       NCON(ID,N) = 6
C
C
   LEFT COEFFICIENT
       COEF(I,N) = SYY
       COEF(IB,N) = EYY
       COEF(ID.N) = CYY
       N = N + 1
       NCON(I,N) = 10*(I-1)+1
       NCON(IB.N) = 10*(IB-1)+1
       NCON(ID,N) = 10*(ID-1)+1
(`
    RIGHT COEFFICIENT
       COFF(I,N) = SYY
       COEF(IB.N) = FYY
       COEF(ID,N) = CYY
       N = N + 1
       NCON(I,N) = 10*(I+1)+I
       NCON(IB,N) = 10*(IB+1)+1
       NCON(ID.N) = 10*(ID+1)+1
(
C
    FRONT COEFFICIENT
       COEF(I.N) = SXX
       COEF(IB,N) = EXX
       COEF(ID,N) = CXX
       N = N + 1
       NCON(I,N) = 10*(I-NWIDE)+1
       NCON(IB,N) = 10*(IB-NWIDE)+1
       NCON(ID,N) = 10*(ID-NWIDE)+1
C.
    BACK COEFFICIENT
       COEF(I,N) = SXX
       COEF(IB.N) = EXX
       COEF(ID,N) = CXX
       N = N + 1
       NCON(LN) = 10*(I+NWIDE)+1
       NCON(IB,N) = 10*(IB+NWIDF)+1
       NCON(ID,N) = 10*(ID+NWIDE)+1
    TOP COEFFICIENT
       COEF(LN) = SZT
       COEF(IB,N) = SZE
       COEF(ID,N) = EZC
       N = N + 1
       NCON(I.N) = 7511
       NCON(IB.N) = 10*I+1
       NCON(ID.N) = 10*IB+1
   BOTTOM COEFFICIENT
```

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```
COEF(I,N) = SZI:
     COEF(IB.N) = EZC
     COEF(ID.N) = CZB
     N = N + 1
     NCON(LN) = 10^{\circ}(1+NPL)+1
     NCON(IB,N) = 10^{\circ}(IB + NPL) + 1
     NCON(ID.N) = 7561
C
   HEAT INPUT
     IF (HEAT(III(1)JH(1)).NE.0.0) THEN
       COEF(LN) = HFAT(H(L)JH(L))
       N = N + 1
       NCON(I,N) = 9991
     ENDIF
(,
    ENDIF
90
  CONTINUE
C
C
(
C DETERMINE CONSTANTS FOR CARRIER COEFFICIENTS
C
C EAR SIZE
  FAR = (CL-SL)/2
(
  *************EAR COLFFICIENTS IF NECCESARY*******************
(
   IF (EAR.NE.0.0) THEN
(
C. DETERMINE THE NECCESARY CONSTANTS
C CONSTANTS FOR EAR NODES
C. LEFT OR RIGHT TO EXTERIOR
  CYLRE = 2*KC*EAR*DELC/DELY
(
 LEFT OR RIGHT TO INTERIOR
  CYYE = EAR*KC*DELC/DELY
(
C
 FRONT OR BACK TO EXTERIOR
  CXFBE = 2*DELY*KC*DELC/EAR
 FRONT OR BACK TO INTERIOR
  CXXE = DELY*DELC*KC*2/(EAR+DELX)
(
 TOP OR BOTTOM
  CZFAR = KC*FAR*DELY*2DELC
(
 DETERMINE LAR COEFFICIENTS
    DO 100.1 = 1.2*NWIDI
     II = 3^{\circ}NPL + I
     N = 1
```

```
C EAR NODE CONNECTIONS
       NCON(IE,N) = 6
(
C CORNERS
       IF ((IH(I).EQ.1.OR.IH(I).EQ.2).AND.(JII(I).EQ.1.OR.JH(I).EQ.
   +NWIDE)) THEN
C. LEFT EDGE
         IF (JH(1).EQ.1) THEN
(
      LEFT COEFFICIENT
          COEF(IE.N) = CYLRE
          N = N + 1
          NCON(HEN) = 7551
(
      RIGHT COEFFICIENT
          COFF(IF.N) = CYYE
          N = N + 1
          NCON(1E,N) = 10*(1E+1)+1
         FLSEIF (JH(1).EQ.NWIDE) THEN
C RIGHT EDGE
      LEFT COEFFICIENT
C
          COEF(IE,N) = CYYE
          N = N + 1
          NCON(IE,N) = 10*(IE-1)+1
(
      RIGHT COEFFICIENT
          COEF(IE,N) = CYLRE
          N = N + 1
          NCON(IE,N) = 7541
        ENDIF
C
C FRONT, BACK AND TOP COEFFICIENTS
C FRONT EDGE
        IF (IH(I).EQ.1) THEN
C
      FRONT COEFFICIENT
          COEF(IE.N) = CXFBE
           N = N + 1
          NCON(IE,N) = 75^{\circ}1
(
      BACK COLFFICIENT
          COFF(IE.N) = CXXE
          N = N + 1
          NCON(IF.N) = 10*(2*NPL+JH(1))+1
(
      TOP COEFFICIENT
          COEF(IE,N) = CZEAR
          N = N + 1
          NCON(1E.N) = 7521
C BACK EDGE
         ELSEIF (IH(I).EQ.2) THEN
C
      FRONT COEFFICIENT
          COEF(IE,N) = CXXE
           N = N + 1
           NCON(IE,N) = 10*(3*NPL-NWIDE+JH(I))+1
C
      BACK COEFFICIENT
          COEF(IE,N) = CXFBE
          N = N + 1
          NCON(IE,N) = 7531
(`
      TOP COEFFICIENT
          COEF(IE.N) = CZEAR
```

(

```
N = N + 1
          NCON(IE,N) = 7531
        ENDIF
(
C BOTTOM COEFFICIENT
          COEF(IE,N) = CZEAR
          N = N + 1
          NCON(IE,N) = 7561
C
  EAR NODES ON FRONT EDGE EXCLUDING CORNERS
       ELSEIF ((IH(I).EQ.1).AND.(JH(I).NE.1.OR.JH(I).NE.NWIDE))THEN
C
      LEFT COEFFICIENT
        COEF(IE,N) = CYYE
        N = N + 1
         NCON(IE,N) = 10*(3*NPL+JH(I)-1)+1
(
      RIGHT EDGE
        COEF(IE.N) = CYYF
        N = N + 1
         NCON(IE,N) = 10^{*}(3*NPL+JH(1)+1)+1
(
      FRONT COEFFICIENT
        COEF(IE.N) = CXFBF
        N = N + 1
        NCON(IE,N) = 7521
(
      BACK COEFFICIENT
        COEF(IE,N) = CXXE
        N = N + 1
        NCON(IE,N) = 10*(2*NPL+JH(I))+1
(
     TOP COEFFICIENT
        COEF(IE.N) = CZEAR
        N = N + 1
        NCON(IE,N) = 7521
\mathbf{C}
      BOTTOM COEFFICIENT
        COEF(IE,N) = CZEAR
         N = N + 1
        NCON(IE,N) = 7561
  EAR NODES ON BACK EDGE EXCLUDING CORNERS
       ELSEIF ((IH(I).EQ.2).AND.(JH(I).NE.1.OR.JH(I).NE.NWIDE))THEN
     LEFT COEFFICIENT
        COEF(IE.N) = CYYE
         N = N + 1
         NCON(IE,N) = 10*(3*NPL+NWIDE+JH(I)-1)+1
(
      RIGHT COEFFICIENT
        COEF(IE,N) = CYYE
        N = N + 1
         NCON(HE,N) = 10*(3*NPL+NWIDE+JH(I)+1)+1
     FRONT COEFFICIENT
(
        COEF(IE,N) = CXXF
        N = N + 1
        NCON(IE.N) = 10*(3*NPL-NWIDE+JH(I))+1
(
      BACK COEFFICIENT
        COEF(IE,N) = CXFBE
         N = N + 1
         NCON(IE,N) = 7531
•
      TOP COEFFICIENT
```

```
COEF(IE.N) = CZEAR
        N = N + 1
        NCON(IE.N) = 7531
     BOTTOM COEFFICIENT
        COFF(IE.N) = CZEAR
        N = N + 1
        NCON(IF,N) = 7561
      FNDIF
100
     CONTINUE
(
C
(
\mathbf{C}
\mathbf{C}
C
     DO 101 I = 1, NPL
      N = 1
      IC = 2*NPL + I
C
C
   C^*
C
      IIF((IH(I).EQ.1.OR.IH(I).EQ.NDEEP).AND.(JH(I).EQ.1.OR.JH(I)
   +.EQ.NWIDE)) THEN
(
(
  NUMBER OF CONNECTIONS PER NODE
      NCON(IC,N) = 6
C
  LEFT AND RIGHT COEFFICIENTS DEPENDING ON WHICH EDGE
(
(
     LEFT EDGE
        IF (JH(I).EQ.1) THEN
\mathbf{C}
     LEFT COEFFICIENT
         COEF(IC.N) = CYLR
          N = N + 1
          NCON(IC,N) = 7551
\mathbf{C}
     RIGHT COEFFICIENT
         COEF(IC.N) = CYY
          N = N + 1
          NCON(IC,N) = 10*(IC+1)+1
\mathbf{C}
     RIGHT EDGE
        ELSEIF (JH(I).EQ.NWIDE) THEN
(
     LEFT COEFFICIENT
          COEF(IC,N) = CYY
          N = N + 1
          NCON(IC,N) = 10*(IC-1)+1
     RIGHT COEFFICIENT
         COEF(IC.N) = CYLR
          N = N + 1
          NCON(IC,N) = 7541
        ENDIF
(
  FRONT AND BACK COFFFICIENTS DEPENDING ON WHICH EDGE
(
     FRONT EDGE
        IF (IH(I).EQ.1) THFN
```

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```
\mathbf{C}
      FRONT COEFFICIENT
           COEF(IC,N) = CXXE
           N = N + 1
           NCON(IC,N) = 10*(3*NPL+JH(I))+1
(
      BACK COEFFICIENT
           COEF(IC,N) = CXX
           N = N + 1
           NCON(IC.N) = 10*(IC+NWIDE)+1
(
      BACK EDGE
         ELSETE (III(1),FQ.NDELP), THEN
(
      FRONT COFFFICIENT
           COEF(IC,N) = CXX
           N = N + 1
           NCON(IC.N) = 10*(IC-NWIDE)+1
(
      BACK COEFFICIENT
           COEF(IC.N) = CXXE
           N = N + 1
           NCON(IC.N) = 10*(3*NPL+NWIDE+JH(I))+1
         ENDIF
\mathbf{C}
\mathbf{C}
      TOP COEFFICIENTS
         COEF(IC,N) = EZC
         N = N + 1
         NCON(IC.N) = 10*(IC-NPL)+1
\mathbf{C}
      BOTTOM COEFFICIENT
         COEF(IC,N) = CZB
         N = N + 1
         NCON(IC,N) = 7561
(
C
(**
     ********************FRONT AND BACK EDGES EXCLUDING CORNERS**********
(
(
       \verb+LSEIF+ ((\verb+IH(I).FQ.1.OR.IH(I).EQ.NDEEP+).AND.(\verb+JH(I).NE.1.OR.
   +JH(I).NE.NWIDE)) THEN
C
(
        NUMBER OF CONNECTIONS PER NODE
       NCON(IC.N) = 6
(
(
      LEFT COEFFICIENT
         COEF(IC.N) = CYY
         N = N + 1
         NCON(IC,N) = 10*(IC-1)+1
(
(
      RIGHT COEFFICIENT
         COEF(IC,N) = CYY
         N = N + 1
         NCON(IC,N) = 10*(IC+1)+1
C
\mathbf{C}
      FRONT AND BACK COEFFICIENTS DEPENDING ON WHICH EDGE
\mathbf{C}
      FRONT EDGE
         IF (IH(I).EQ.1) THEN
(
      FRONT COEFFICIENT
           COEF(IC,N) = CXFBE
           N = N + 1
```

```
NCON(IC.N) = 10*(3*NPL+JH(I))+1
(
     BACK COEFFICIENT
          COEF(IC,N) = CXX
          N = N + 1
          NCON(IC,N) = 10*(IC+NWIDE)+1
C
     BACK EDGE
        ELSEIF (IH(I).EQ.NDEEP) THEN
C
     FRONT COEFFICIENT
          COEF(IC.N) = CXX
          N = N + 1
          NCON(IC.N) = 10*(IC-NWIDE)+1
\mathbf{C}
     BACK COEFFICIENT
          COEF(IC,N) = CXFBE
          N = N + 1
          NCON(IC.N) = 10*(3*NPL+NWIDE+JH(1))+1
        ENDIF
(
(
      TOP COEFFICIENT
        COEF(IC.N) = EZC
        N = N + 1
        NCON(IC.N) = 10*(IC-NPL)+1
(,
(
     BOTTOM COEFFICIENT
        COEF(IC.N) = CZB
        N = N + 1
        NCON(IC.N) = 7561
      ENDIF
101
     CONTINUE
C*****FRONT AND BACK EDGE CARRIER COEFFICIENT IF NO EAR****
C****
(
   ELSEIF (EAR.EQ.0.0) THEN
     DO 102 I = 1.NPL
      N = 1
      IC = 2*NPL+I
(
(
      II: ((III(I).EQ.I.OR.III(I).EQ.NDEEP).AND.(JII(I).EQ.I.OR.JII(I).
   +EQ.NWIDE() THEN
(
(
  NUMBER OF CONNECTIONS PER NODE
        NCON(IC.N) = 6
(
  LEFT AND RIGHT COEFFICIENTS DEPENDING ON WHICH EDGE
(
(
     LEFT EDGE
        IF (JH(I).EQ.1) THEN
C
     LEFT COEFFICIENT
          COLF(IC.N) = CYLR
          N = N + 1
          NCON(IC,N) = 7551
C
     RIGHT COEFFICIENT
          COEF(IC.N) = CYY
```

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```
N = N + 1
          NCON(IC,N) = 10^{8}(IC+1)+1
(
     RIGHT FDGF
        FLSEIF (JH(I).FQ.NWIDF) THEN
(
     LEFT COEFFICIENT
          COEF(IC.N) = CYY
          N = N + 1
          NCON(IC.N) = 10*(IC-1)+1
(
     RIGHT COEFFICIENT
          COEF(IC,N) = CYLR
          N = N + 1
          NCON(IC.N) = 7541
        ENDIF
(
C FRONT AND BACK COEFFICIENTS DEPENDING ON WHICH EDGE
C
     FRONT EDGE
        IF (IH(I).EQ.1) THEN
C
     FRONT COEFFICIENT
          COEF(IC,N) = CXFB
          N = N + 1
          NCON(IC,N) = 7521
(
     BACK COFFFICIENT
          COEF(IC,N) = CXX
          N = N + 1
          NCON(IC.N) = 10*(IC+NWIDF)+1
(
     BACK EDGE
        FLSEIF (IH(I).EQ.NDEEP) THEN
(
     FRONT COEFFICIENT
          COEF(IC.N) = CXX
          N = N + 1
          NCON(IC,N) = 10*(IC-NWIDE)+1
(
     BACK COEFFICIENT
          COEF(IC.N) = CXFB
          N = N + 1
          NCON(IC,N) = 7531
        ENDIF
(
(
     TOP COEFFICIENT
        COEF(IC,N) = EZC
         N = N + 1
        NCON(IC,N) = 10*(IC-NPL)+1
(
     BOTTOM COEFFICIENT
        COEF(IC,N) = CZB
        N = N + 1
        NCON(IC,N) = 7561
(`
(
(`
      FLSEIF ((IH(I).EQ.I.OR.IH(I).EQ.NDFEP).AND.(JH(I).NE.I.OR.
   +JH(I).NE.NWIDE()) THEN
C. NUMBER OF CONNECTIONS PER NODE
        NCON(IC.N) = 6
     J.FFI COFFFICIENT
```

```
COEF(IC,N) = CYY
        N = N + 1
         NCON(IC.N) = 10*(IC-1)+1
(
     RIGHT COEFFICIENT
        COEF(IC,N) = CYY
         N = N + 1
         NCON(IC.N) = 10*(IC+1)+1
C
     FRONT AND BACK COEFFICIENTS DEPENDING ON WAICH EDGE
(
(
     FRONT FDGF
         IF (IH(I),EQ.1) THEN
     FRONT COEFFICIENT
          COEF(IC.N) = CXFB
          N = N + 1
          NCON(IC,N) = 7521
(
     BACK COEFFICIENT
          COFF(IC.N) = CXX
          N = N + 1
          NCON(IC,N) = 10*(IC+NWIDE)+1
C
     BACK EDGE
        ELSEIF (IH(I).EQ.NDEEP) THEN
C
     FRONT COEFFICIENT
          COEF(IC,N) = CXX
          N = N + 1
          NCON(IC,N) = 10*(IC-NWIDE)+1
C
     BACK COEFFICIENT
          COEF(IC,N) = CXFB
          N = N + 1
          NCON(IC,N) = 7531
        ENDIF
(
(
      TOP COEFFICIENT
        COEF(IC.N) = EZC
         N = N + 1
         NCON(IC,N) = 10^{\circ}(IC-NPL)+1
     BOTTOM COLFFICIENT
        COEF(IC.N) = CZB
        N = N + 1
        NCON(IC.N) = 7561
(
      ENDIF
102
     CONTINUE
   ENDIF
(
C GENERATE DATA FILE VALUES
  TOTAL NODES FOR THIS ASPECT RATIO
   IF (EAR.EQ.0.0) THEN
     COUNT = 3*NPL
     COUNT = 3*NPL+2*NWIDE
   ENDIF
(
C. NUMBER OF CONSTANT TEMPERATURE INPUTS
   CONTEMP = 6
```

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```
C DUMMY VARIABLE
   ZER = 0
\mathbf{C}
C UNITS TO BE USED
   IF (UNITS.EQ.'E') THEN
     USEL = 1
     USEL = 2
   ENDIF
C
C PROBLEM CAPABILITY LINE
C MAXIMUM NODES
   NMAX = 750
C MAXIMUM CONSTANT TEMPERATURES
   TMAX = 50
C NUMBER OF HEATERS
   HTRS = 6
C DATA SETS REQUIRED
   D1 = 2
   D2 = 4
   D3 = 6
   D4 = 0
   D5 = 0
   D6 = 0
   D7 = 0
C
C ACCURACY LINE
C
C ACCURACY BETWEEN ITERATIONS
   ACC = 0.05
C DAMPING VALUE
   DAMP = 0.66667
C MAXIMUM ITERATIONS
   MAXIT = 12
C CONVERGENCE FACTOR
   CONFAC = 0.8
C
\mathbf{C}
(
C. CREATE DATA FILE
   OPEN (3.FILE=NAME.FORM='FORMATTED',ACCESS='DIRECT',RECL=108
   +.STATUS='NEW')
\mathbf{C}
C LINE 1, TITLE
   WRITE(3,909) DATAF
909 FORMAT(1X,A79)
C
C LINE 2, PROBLEM DATA
   WRITE(3,908) COUNT, CONTEMP, ZER, ZER, ZER, ZER, ZER, USEL
908 FORMAT(2X,9(13.5X))
\boldsymbol{C}
C ANALYZER CONTROL LINE
    WRITE(3,907) ZER,ZER,ZER
907 FORMAT(2X,3(I3,5X))
\mathbf{C}
```

```
C PROBLEM CAPABILITY LINE
   WRITE(3,908) NMAX,TMAX.HTRS,D1,D2,D3,D4,D5,D6,D7
\mathbf{C}
C ACCURACY LINE
   WRITE(3,905) ACC, DAMP, MAXIT, CONFAC, IT
905 FORMAT(1X,2(F9.7,1X),14X,I2.1X,F9.7.1X,F9.5)
\mathbf{C}
\mathbf{C}
C CONSTANT TEMPERATURE LINE
   WRITE(3,906) UPRT,FT,BT,RT,LT,LWRT
906 FORMAT(1X,6(F12.9,1X))
C
C COEFFICIENT EQUATIONS
   DO 112 I=1,COUNT
     WRITE(3.910) (NCON(I,J),J=1.8)
910
      FORMAT(I4,3X,7(I4,8X))
     WRITE(3,911) (COEF(I,N),N=1,7)
911
     FORMAT(7(F9.3,3X))
112 CONTINUE
C
C
C
   CALL CLS
   WRITE(*.999) NAME
999 FORMAT(##: THE OUTPUT DATA HAS BEEN PLACED IN A FILE NAMED."
   +,A6)
   END
```

## LIST OF REFERENCES

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